

TALK TO SATELLITE – GROUND STATION TECHNOLOGY

Amita Shrestha
Institute of Communications and Navigation
German Aerospace Center (DLR)



Contents



Institute for Communications and Navigation



The institute is engaged in the design, analysis and realization of systems for communication and navigation for applications in the fields of space, aviation, transport and security.

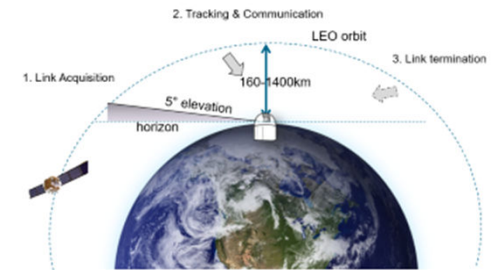
The work ranges from the scientific fundamentals to technology demonstration in a real environment and technology transfer in cooperation with industry.

Optical Satellite Links Department

Free Space Optical Communication



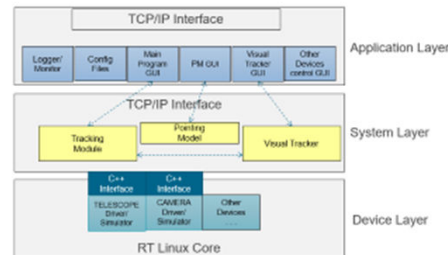
Typical LEO Downlink Scenario



Optical Ground Stations



OGS Software Overview

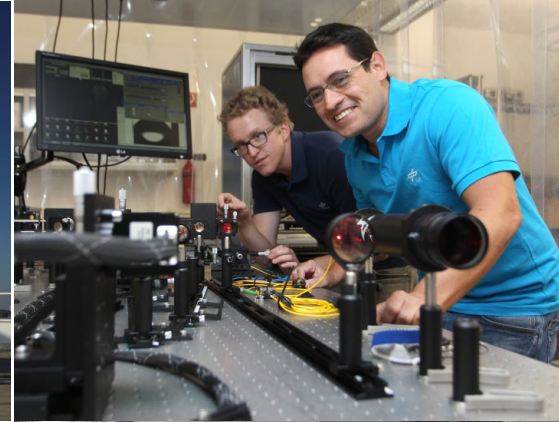
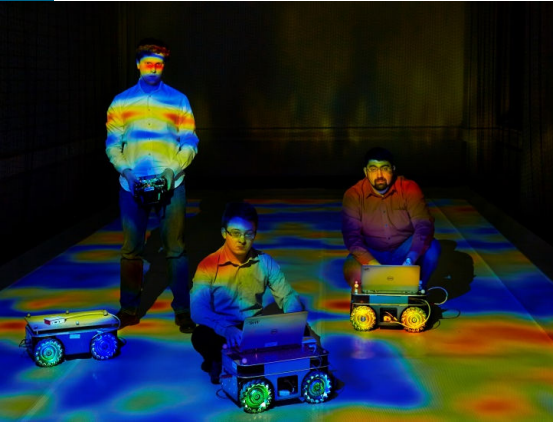


Satellite Based QKD

Why Satellite?
→ bridge large distances

- FSO and QKD enable worldwide fast and secure data communications
- Why QKD over FSO? → use of fiber is range limited → QKD using satellite node with QKD relay protocol
- Combination of FSO and QKD technology in one device → high synergies → lower costs
- QKD schemes: Discrete variable (BB84) or continuous variable (and various others...)

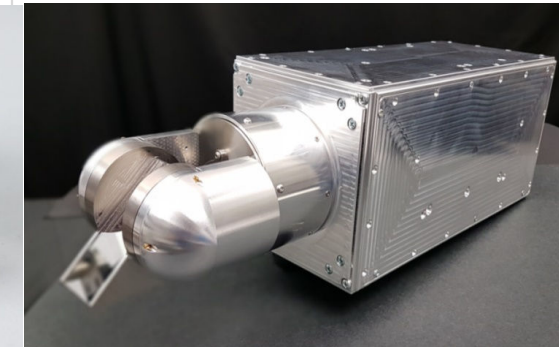
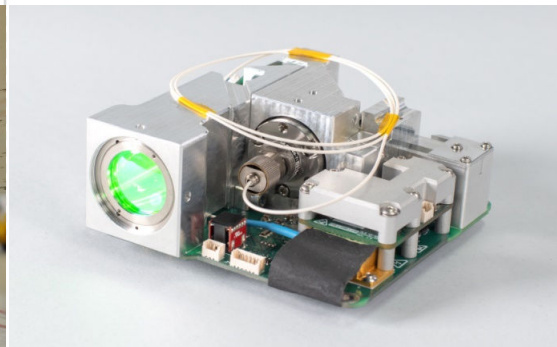
Institute for Communications and Navigation



The institute is engaged in the design, analysis and realization of systems for communication and navigation for applications in the fields of space, aviation, transport and security.

The work ranges from the scientific fundamentals to technology demonstration in a real environment and technology transfer in cooperation with industry.

Optical Satellite Links Department



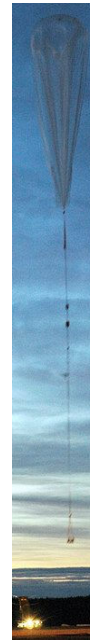
Heritage in Free-Space Optical Transmission



2004: First link from a tethered balloon



2005: First link from the stratosphere, 22 km height
1.25 Gbps, 100 mW



2008: First air-to-ground link
1.25 Gbps, $d=120$ km



2013: First air-to-ground link
Mach 0.7, 1.25 Gbps, $d=60$ km
jointly with ViaLight, Contract by Airbus



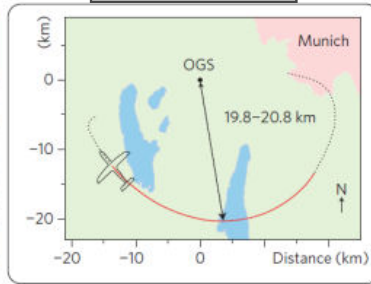
Quantum Key Distribution (QKD) from aircraft to ground



aircraft: Alice



Flight path



Ground Station: Bob

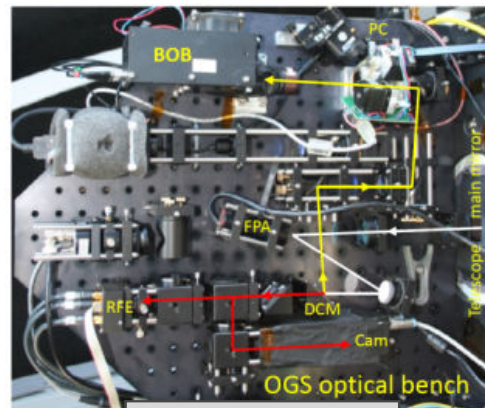


- Investigation of Quantum Key Distribution (QKD) with BB84 protocol
- **Physically secure communications**
- Polarization states of single photons exploited as quantum effect
- Demonstration of QKD between aircraft and ground segment in 2011

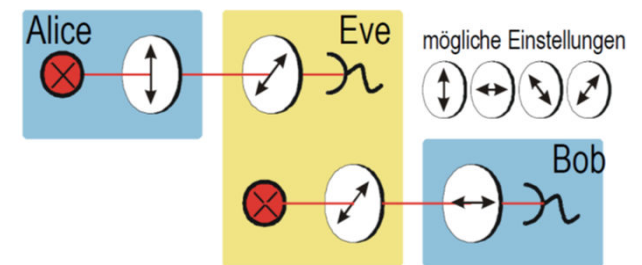
Flight terminal FELTII



Aircraft Optical bench



Ground Station Optical bench

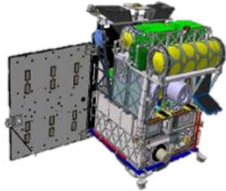


Weinfurter 1998

OSIRIS* Program at IKN



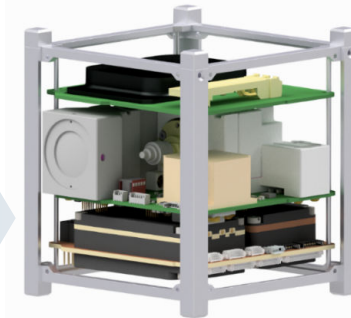
OSIRISv2
Closed-Loop Body Pointing
Data rate: 1 Gbit/s



OSIRIS4CubeSat (O4C)
Active Beam Steering
combined with Body Pointing
Data rate: 100 Mbit/s



CubeISL
O4C Evolution
Inter-Satellite Links: 100 Mbit/s
Downlinks: 1 Gbit/s



2017

2016

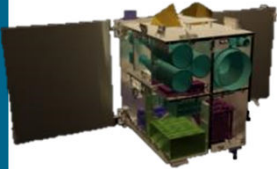
2021

2024

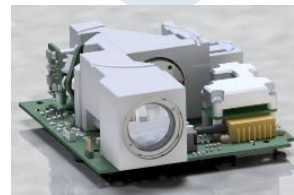
2024

2025

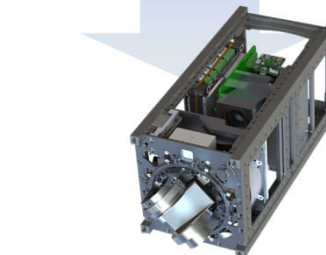
OSIRISv1
Open-Loop Body Pointing
Data rate: 200 Mbit/s



OSIRISv3
Active Beam Steering
with Coarse Pointing Assembly (CPA)
Data rate: 10 Gbit/s



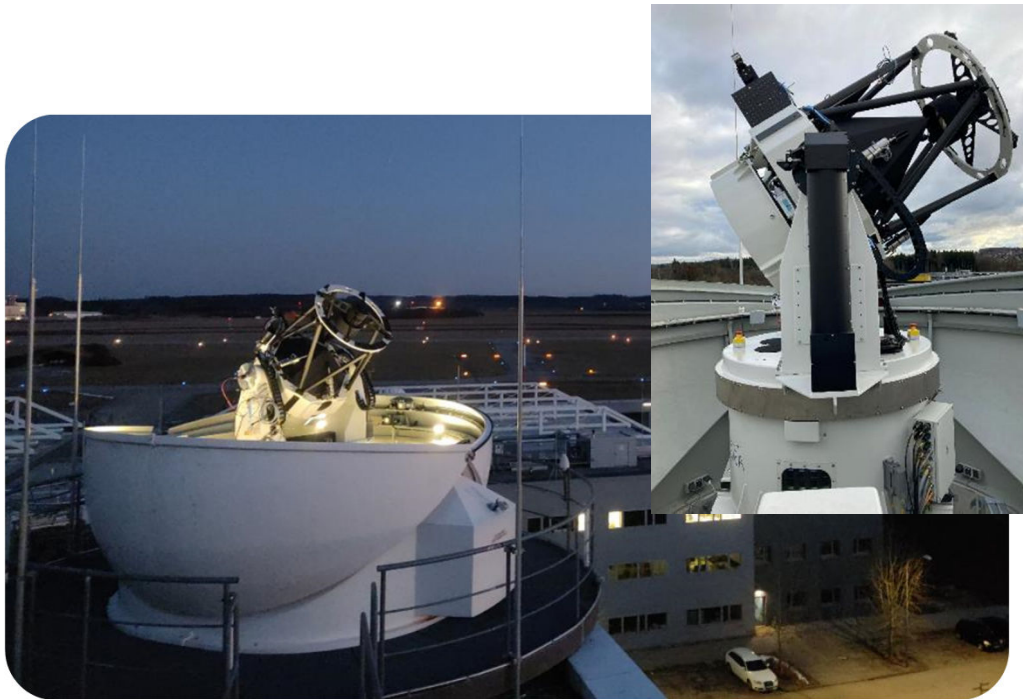
QUBE
O4C Evolution
for QKD-technology
with 1550 nm and 850 nm



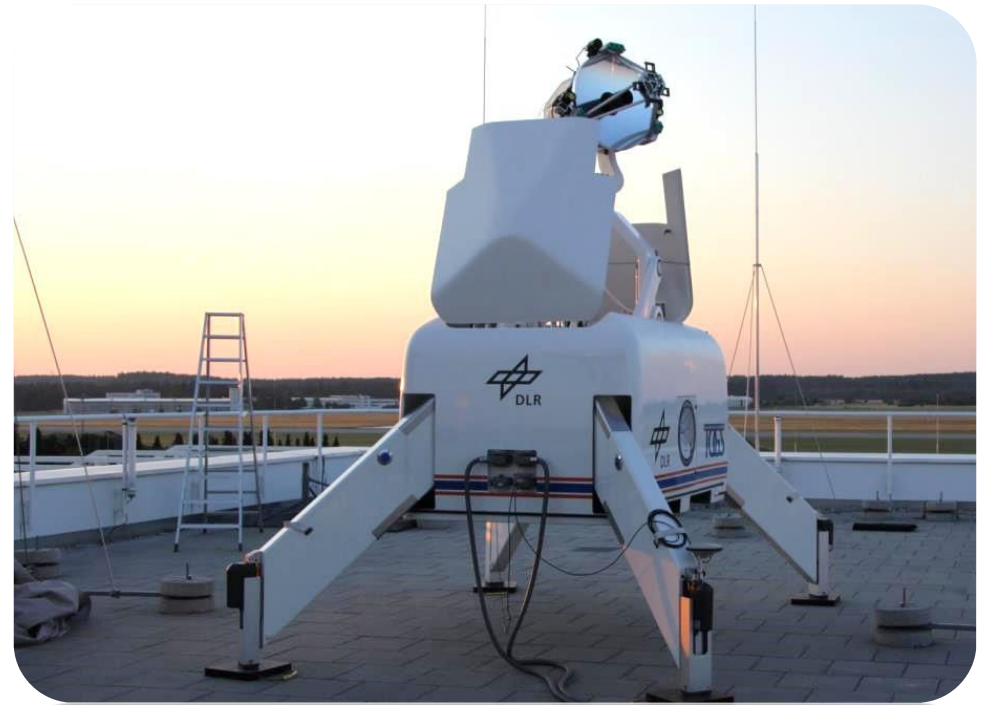
Cube1G
CubeISL Evolution
Active Beam Steering with CPA
Downlinks: 1 Gbit/s

*Optical Space Infrared Downlink System

DLR IKN Optical Ground Stations

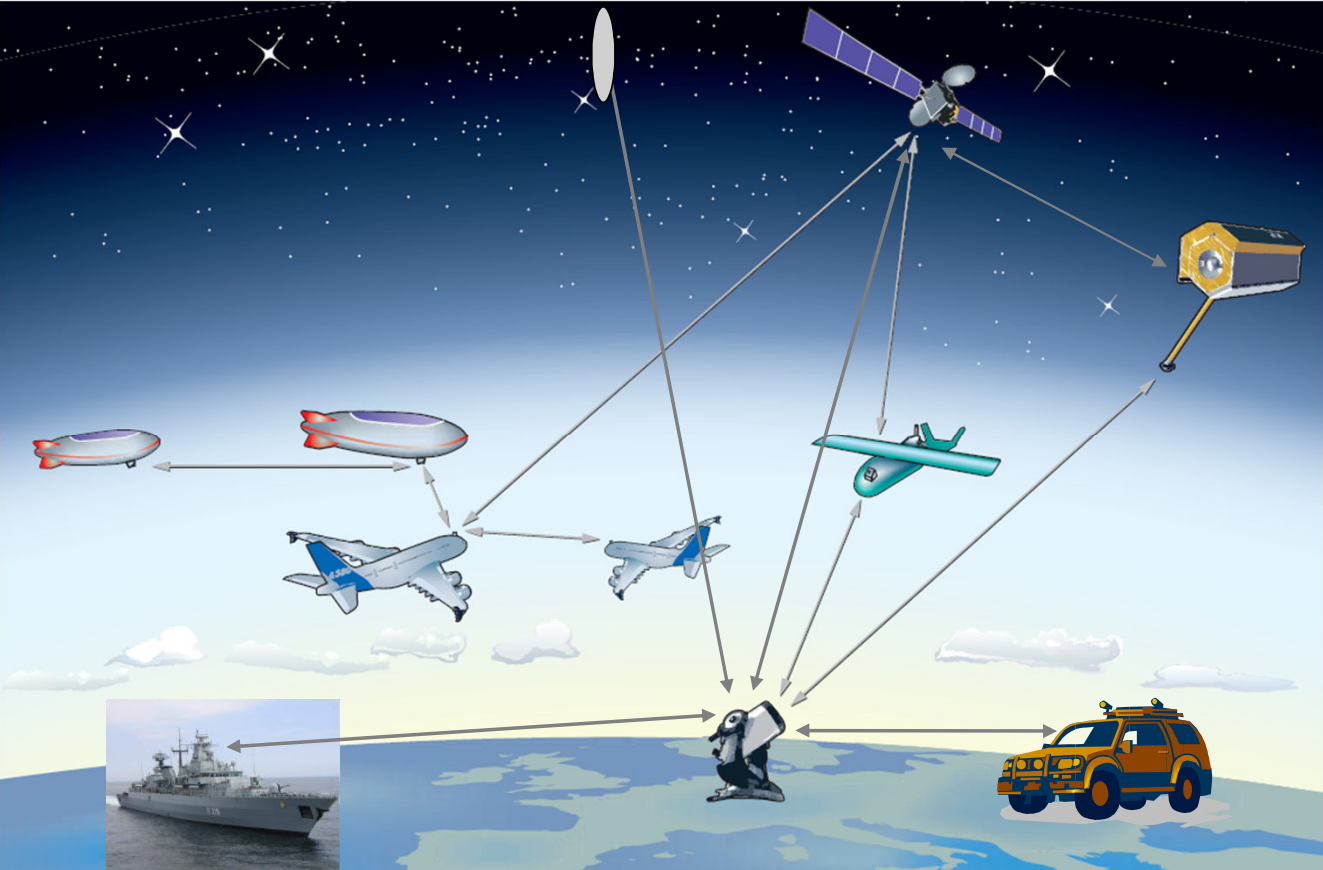


Optical Ground Station Oberpfaffenhon (OGSOP)

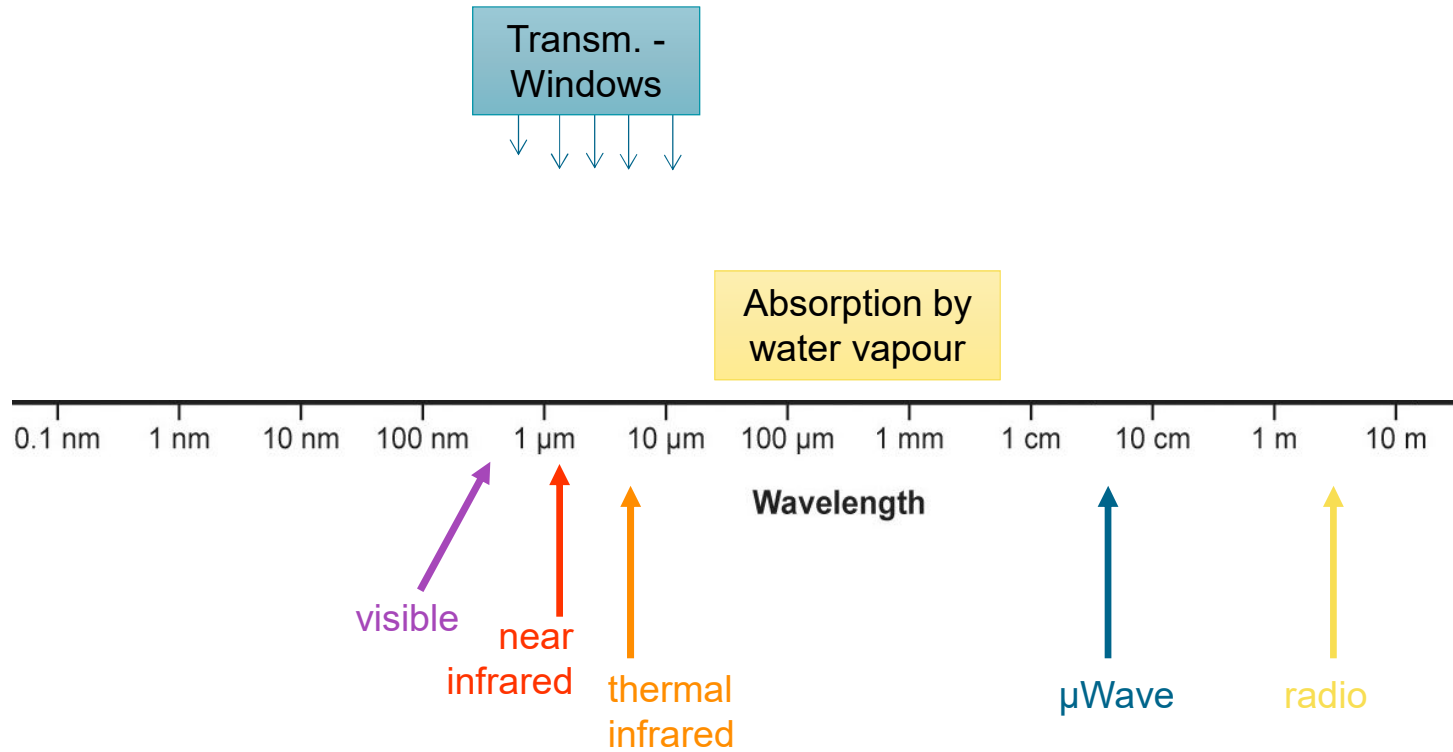


Transportable Optical Ground Station (TOGS)

Free Space Optical Communication



„From RF to Light“ – what wavelengths are used for Space-FSO



Free Space Optical (FSO) Communication



Advantages of FSO:

- High datarate
- Low power consumption
- High security
- No spectrum regulation
- Transmission of Single Photons allow application of Quantum techniques



Typical parameters

- Laser-wavelengths in the near infrared (850nm / 1064nm / 1550nm)
- diffraction limited Tx-divergence: below 1/1000 degree \rightarrow $x \mu\text{rad}$
- datarates from few 100Mbps up to n Tbps

FSO Challenges

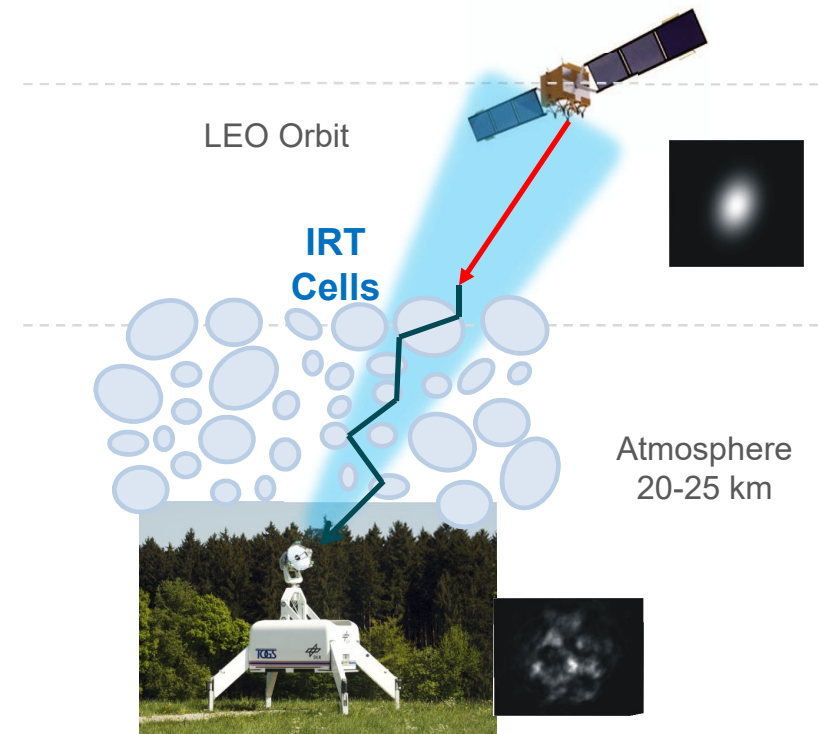


- Challenges:

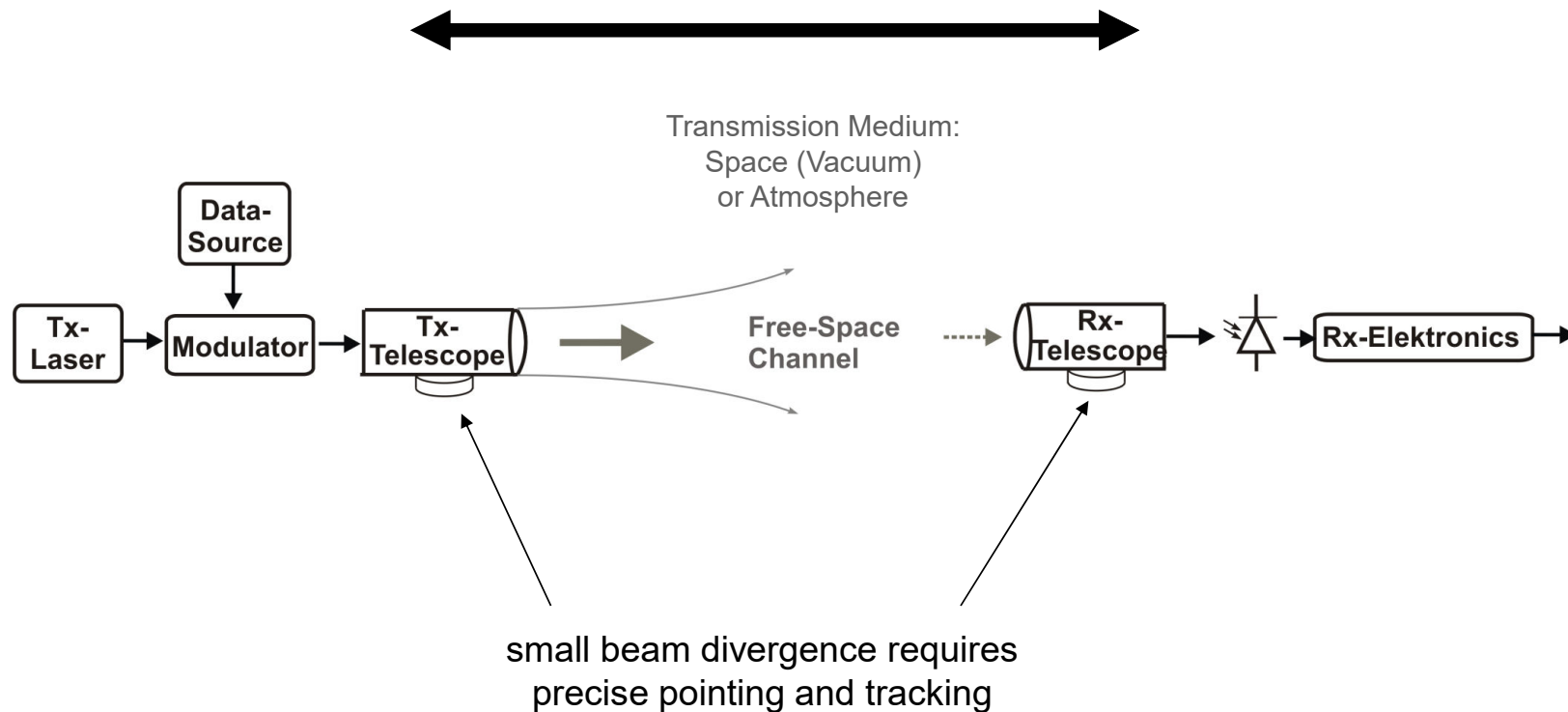
- Turbulent atmosphere: scintillation, fading
- Fog, rain, snow haze: fading, link outages
- Pointing/tracking error : beam wandering

- Mitigation Techniques:

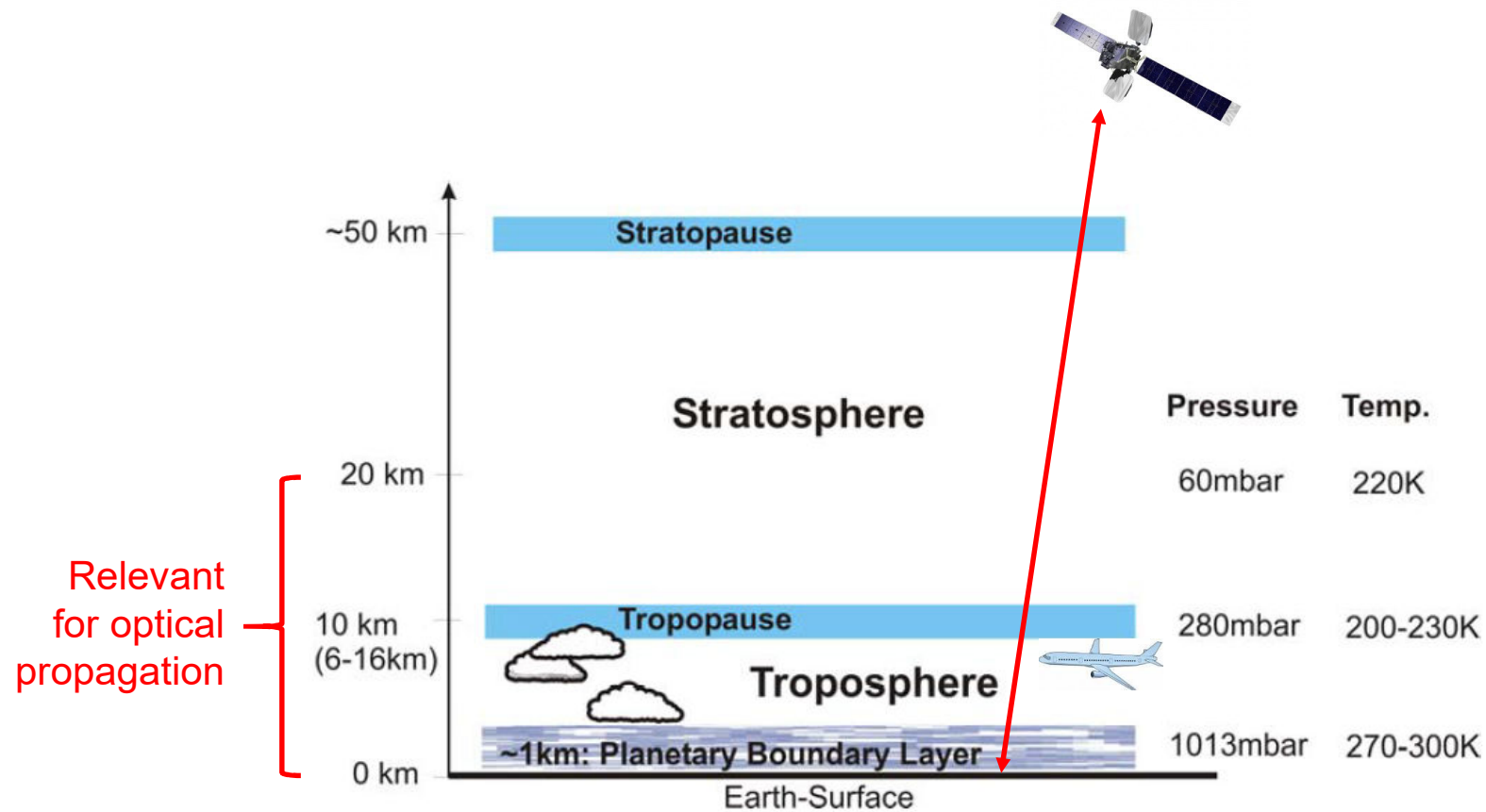
- Aperture averaging using bigger telescopes
- Complex Adaptive optics
- Variable data rate etc.



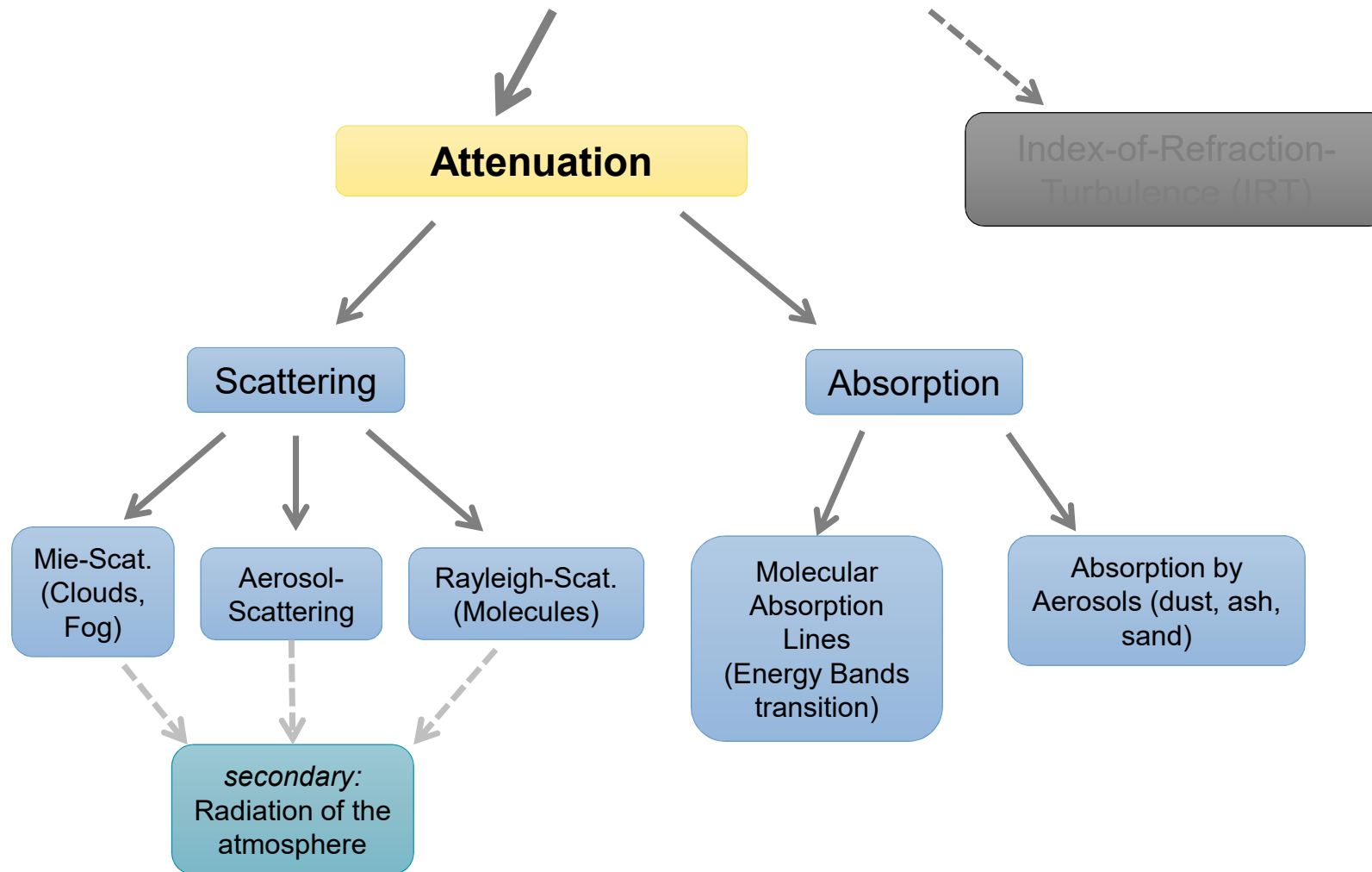
Directed Point-to-Point Links by Modulated Laser Beams → FSOC (Free-Space Optical Communications)



Structure of Earth's Atmosphere



Atmospheric Effects on Optical Signals



Atmospheric transmission: Beer's law



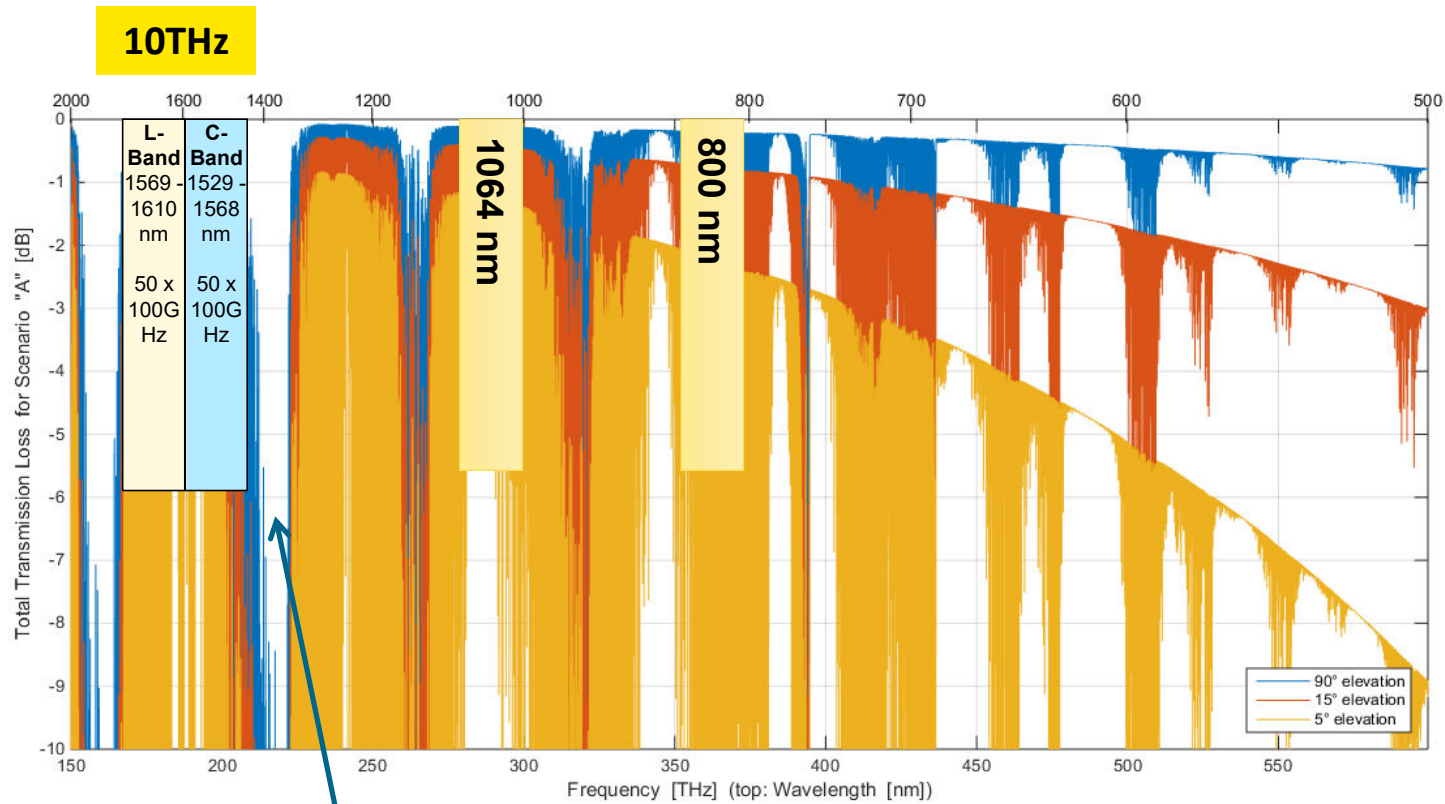
- Calculation of transmission with extinction coefficient (1/km)
- Extinction = Scattering + Absorption
- Accumulated effects considered with one equation → Beer's law
 - Rayleigh scattering
 - Aerosol scattering and absorption
 - Molecular absorption lines

$$I_{out} = I_{in} \cdot \exp\left(-\int_0^L \alpha_{ext}(z, \lambda) dz\right)$$

$$\alpha_{ext}(z, \lambda) = \sum_i \alpha_i(z, \lambda)$$

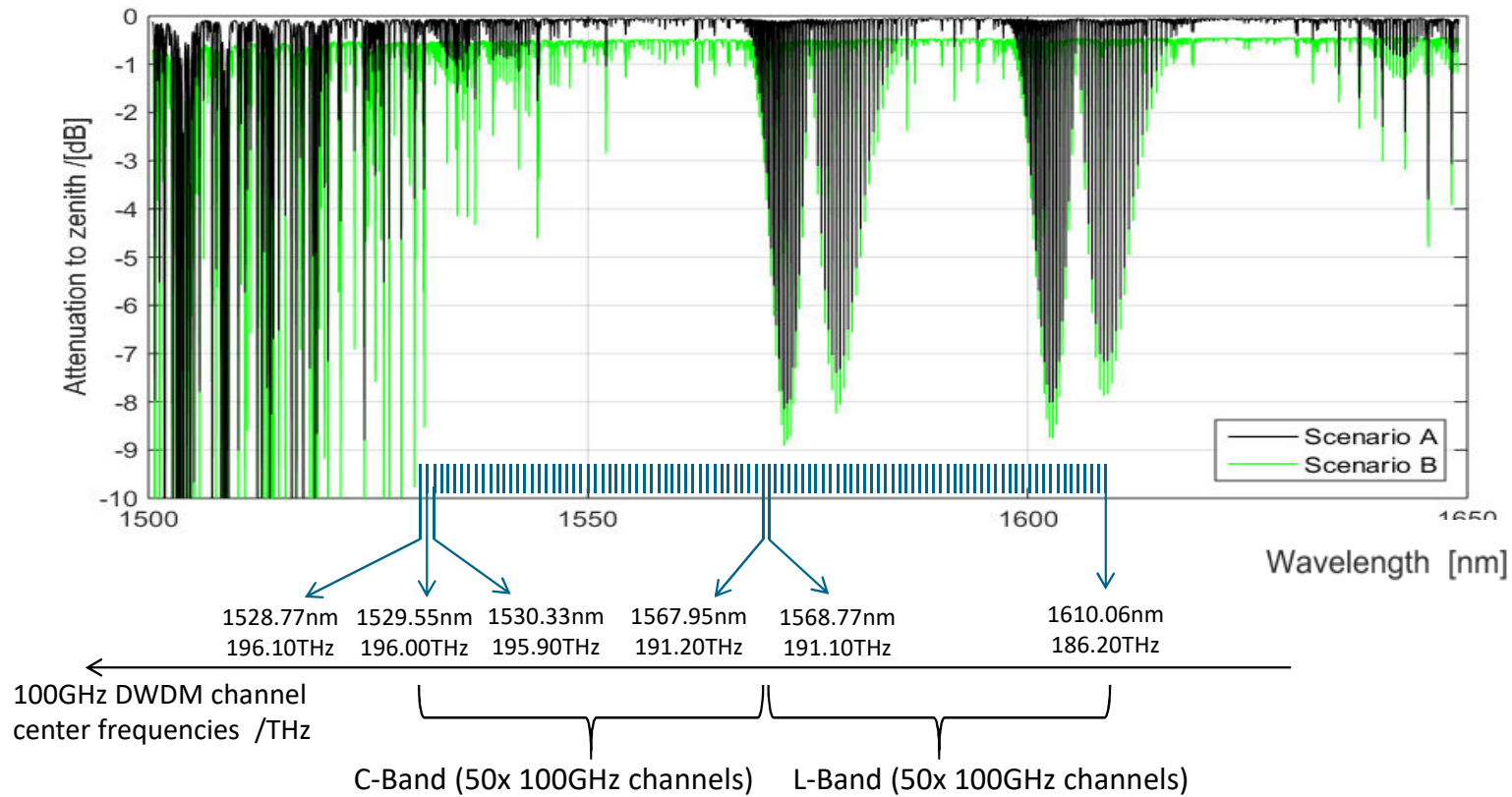
α_{ext}	complete extinction coefficient [km^{-1}]
α_i	various scat./abs. coefficients [km^{-1}]
I	output intensity [W / m^2]
I_0	input intensity [W / m^2]
λ	Wavelength [μm]
L	path length [m]
z	path variable [m]

Available Optical Spectrum between Molecular Absorption Lines



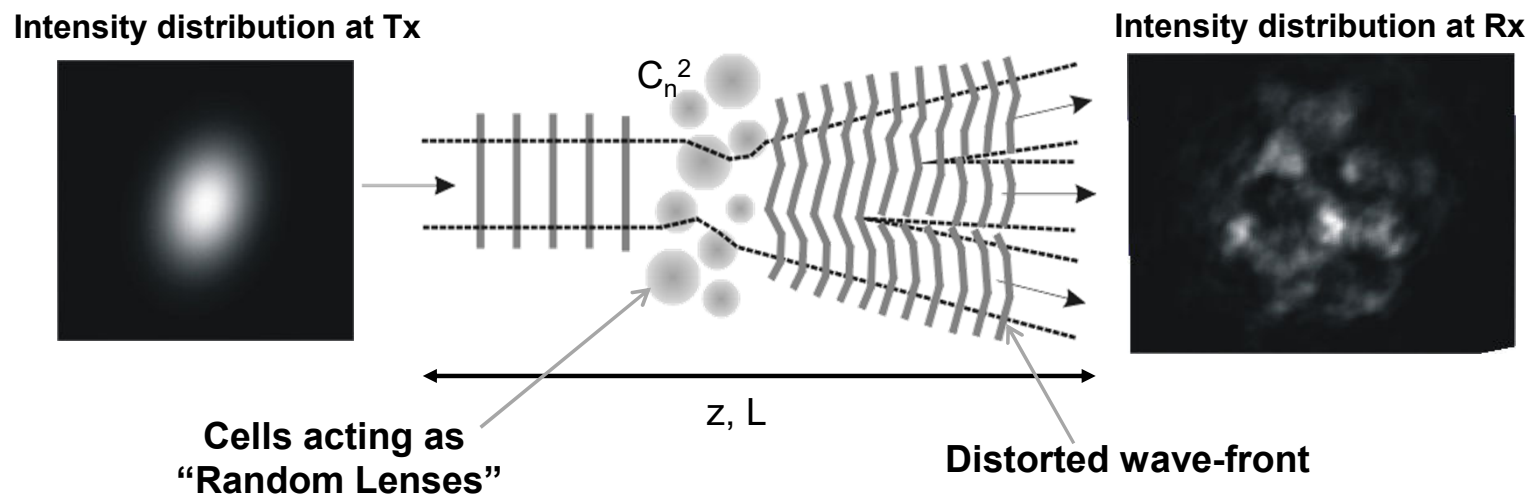
near-IR: absorption mostly by H₂O, in thermal infrared also CO₂

The C- and L-Band DWDM Channels and their transmission (1520nm – 1620nm)



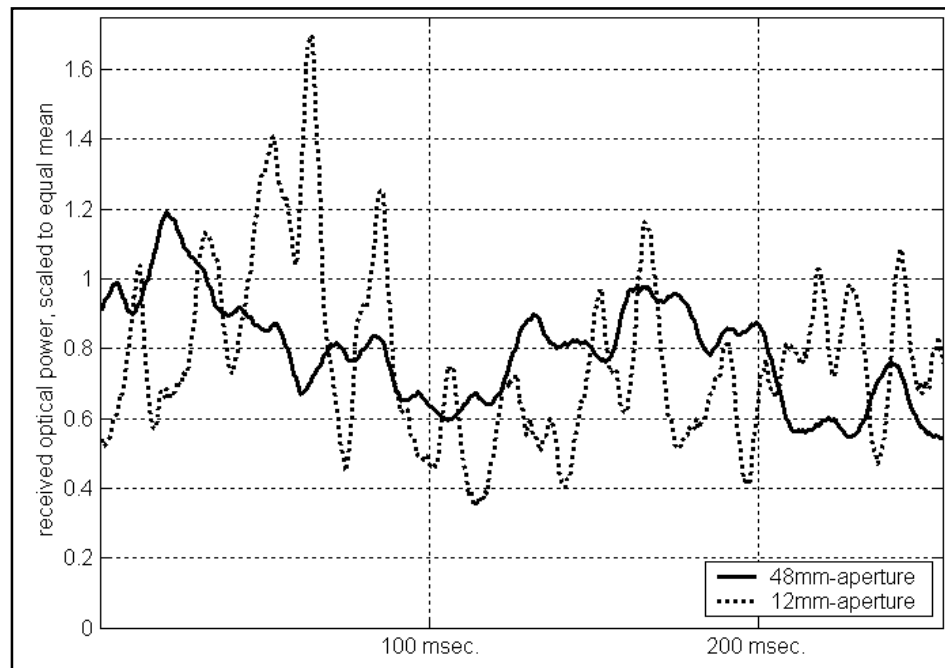
IRT-Scintillation through self-interference

- Beam propagating through optical turbulence → wave-front distortions → several effects
- Severity of the cumulated turbulence and fluctuation regime (weak/moderate/strong)



Aperture Averaging with IRT-Scintillation

depends on ratio intensity-structure vs aperture-diameter



same IRT, two apertures

$$A_f(D) = \frac{\sigma_P^2(D)}{\sigma_I^2}$$

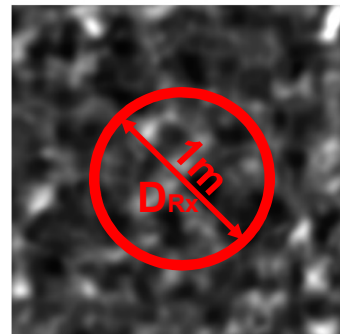
Aperture Averaging with different intensity-cell sizes

Asymmetry in Satellite Uplink vs. Downlink Channel

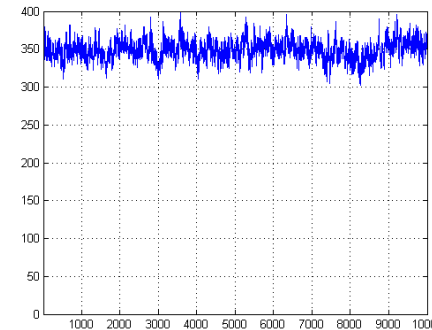


Downlink into
OGS:

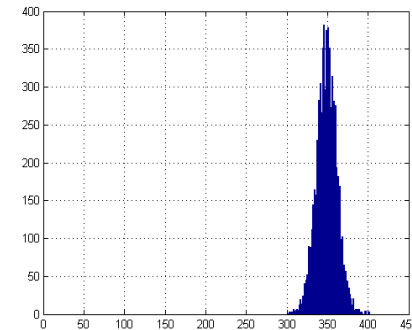
With 1 m aperture
averaging



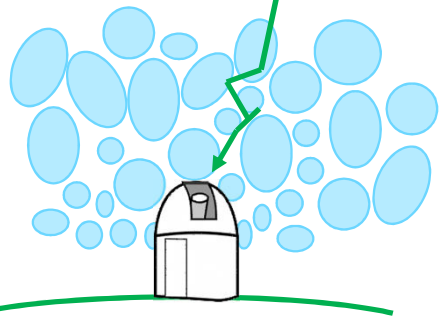
Intensity



1s Rx-Signal

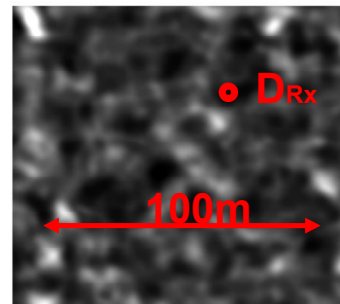


Histogram

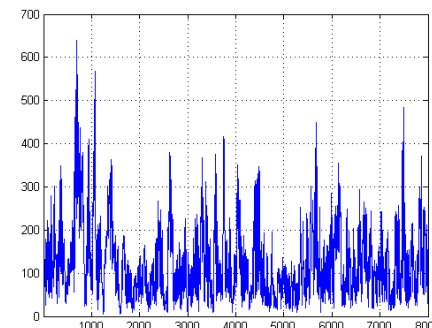


Uplink at Satellite:

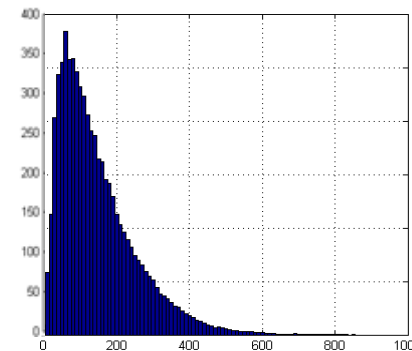
No aperture
averaging
possible



Intensity

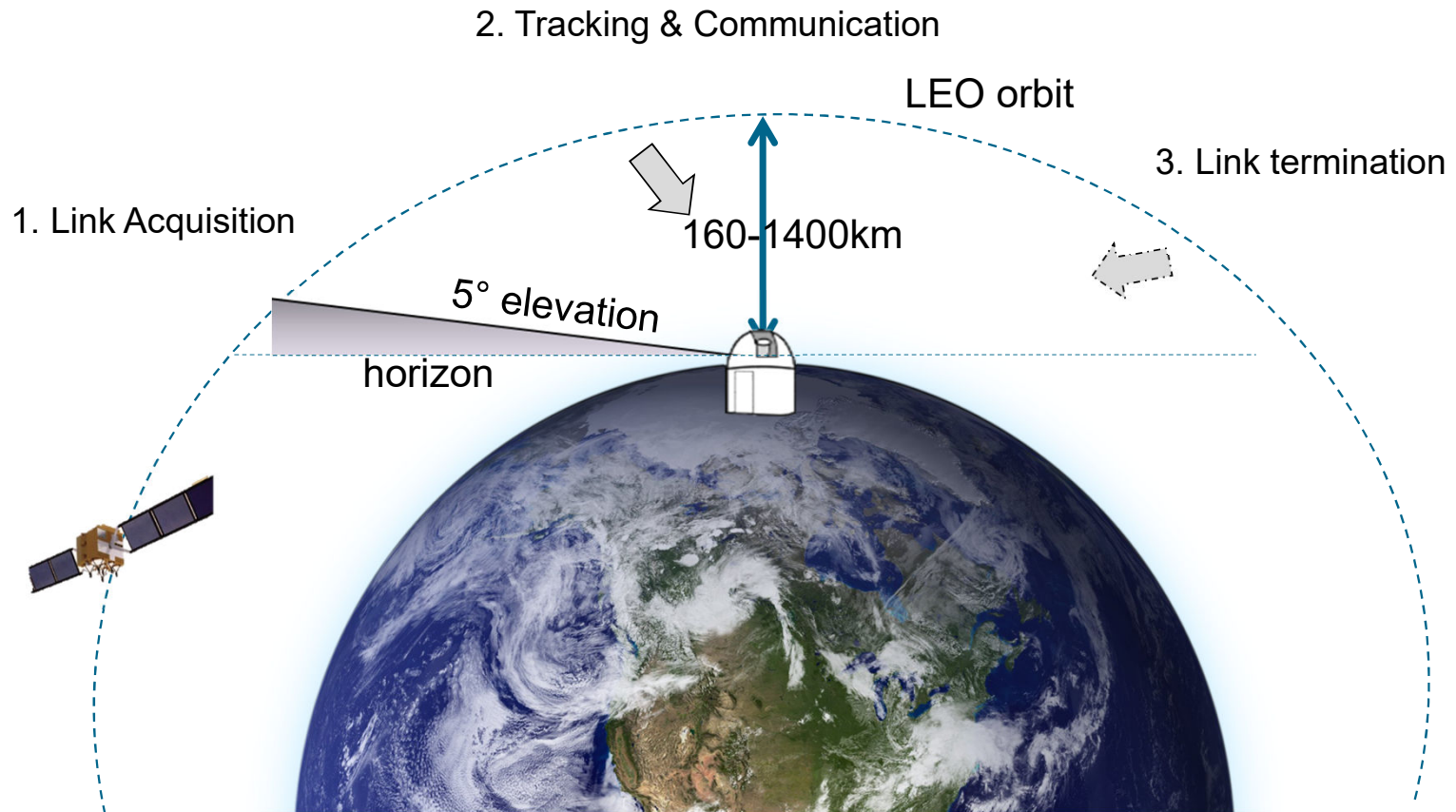


1s Rx-Signal



Histogram

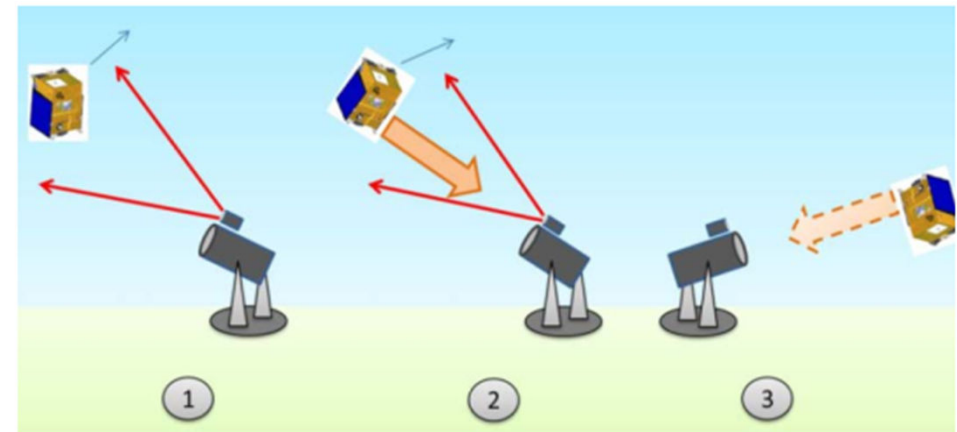
Typical LEO Downlink Scenario



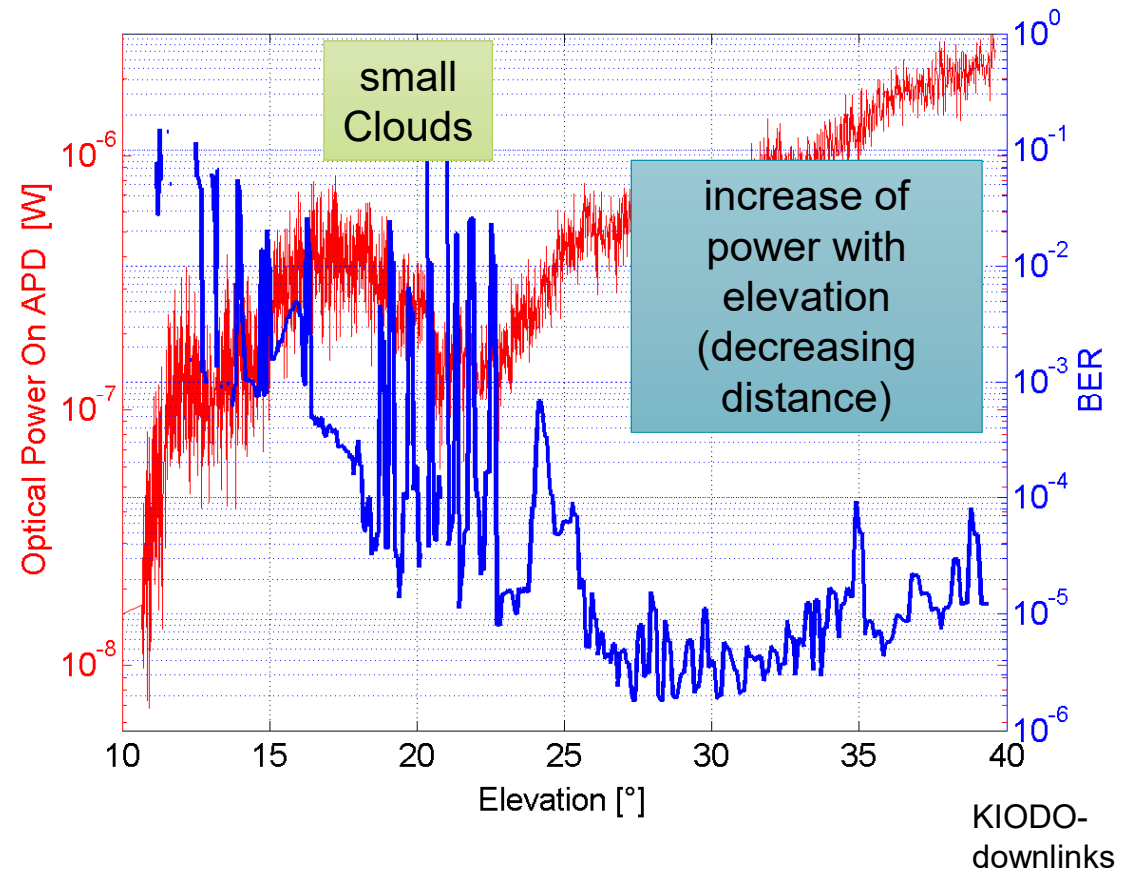
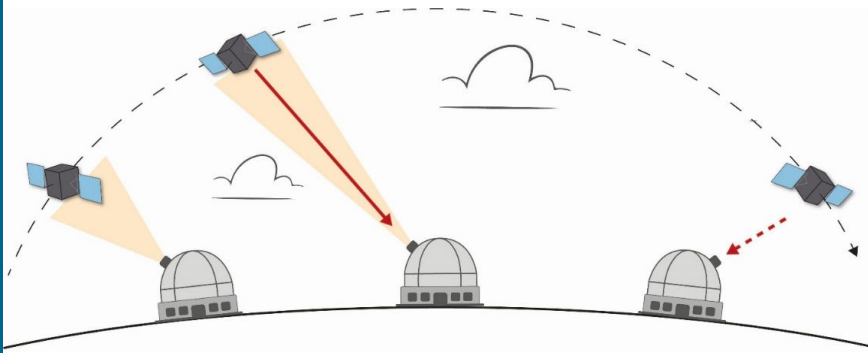
Pointing, Acquisition and Tracking (PAT)



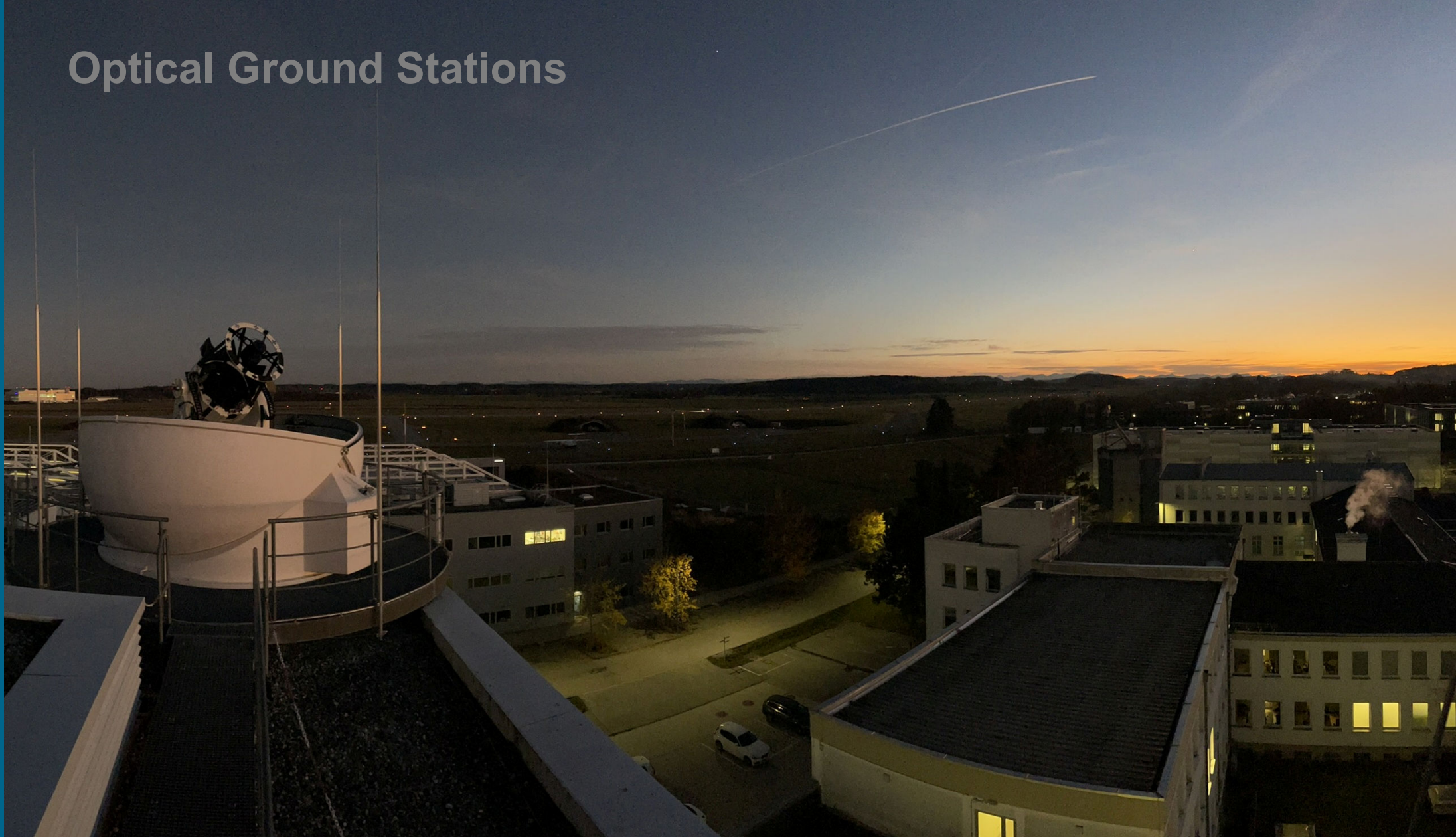
- Pointing
 - Both satellite and GS points towards each other using GPS coordinates, orbit data etc.
- Acquisition
 - Terminals acquires the signal from each other in their acquisition sensor
- Tracking
 - Continuous tracking of the acquired signal with coarse pointing or fine pointing assembly



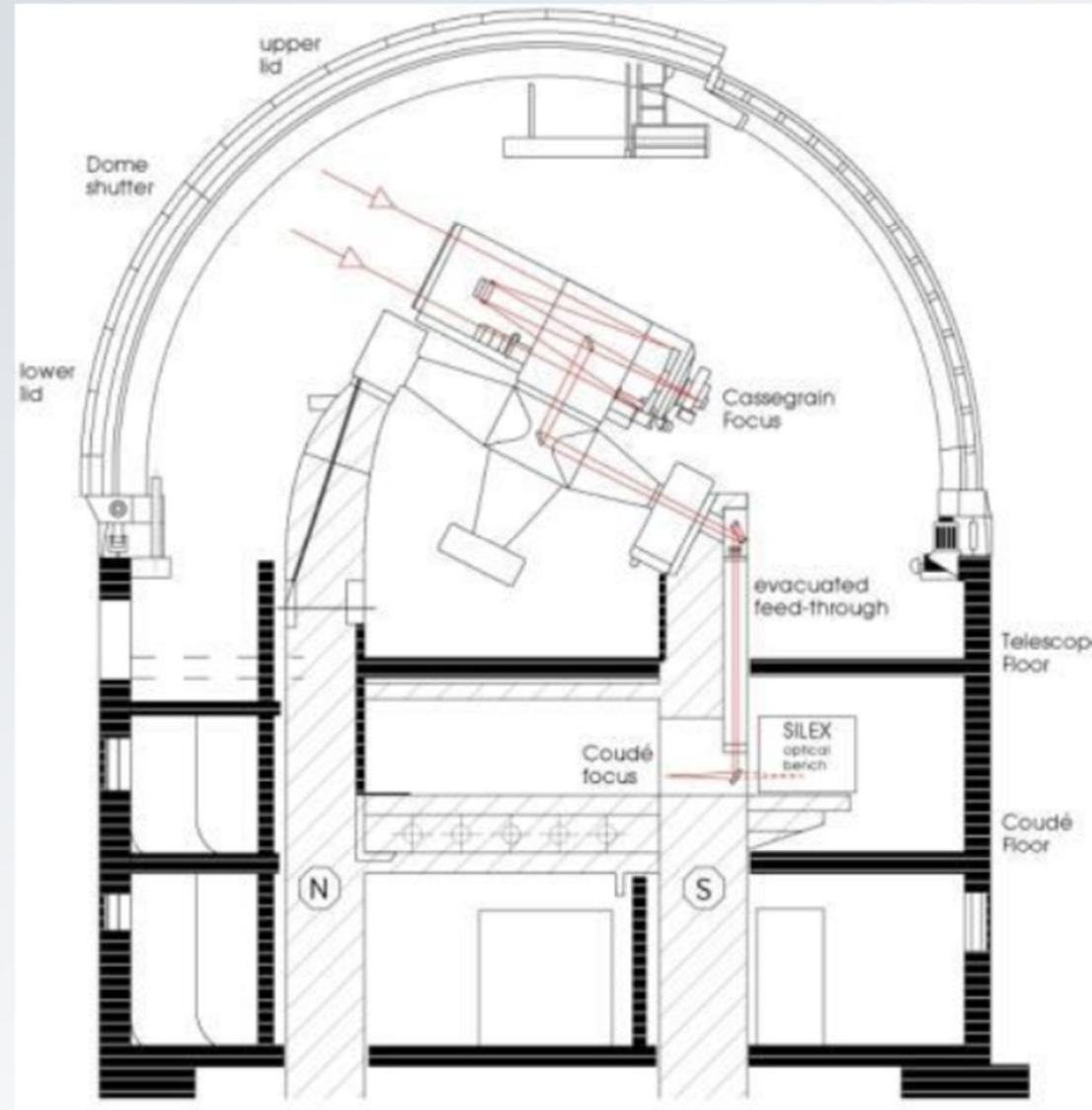
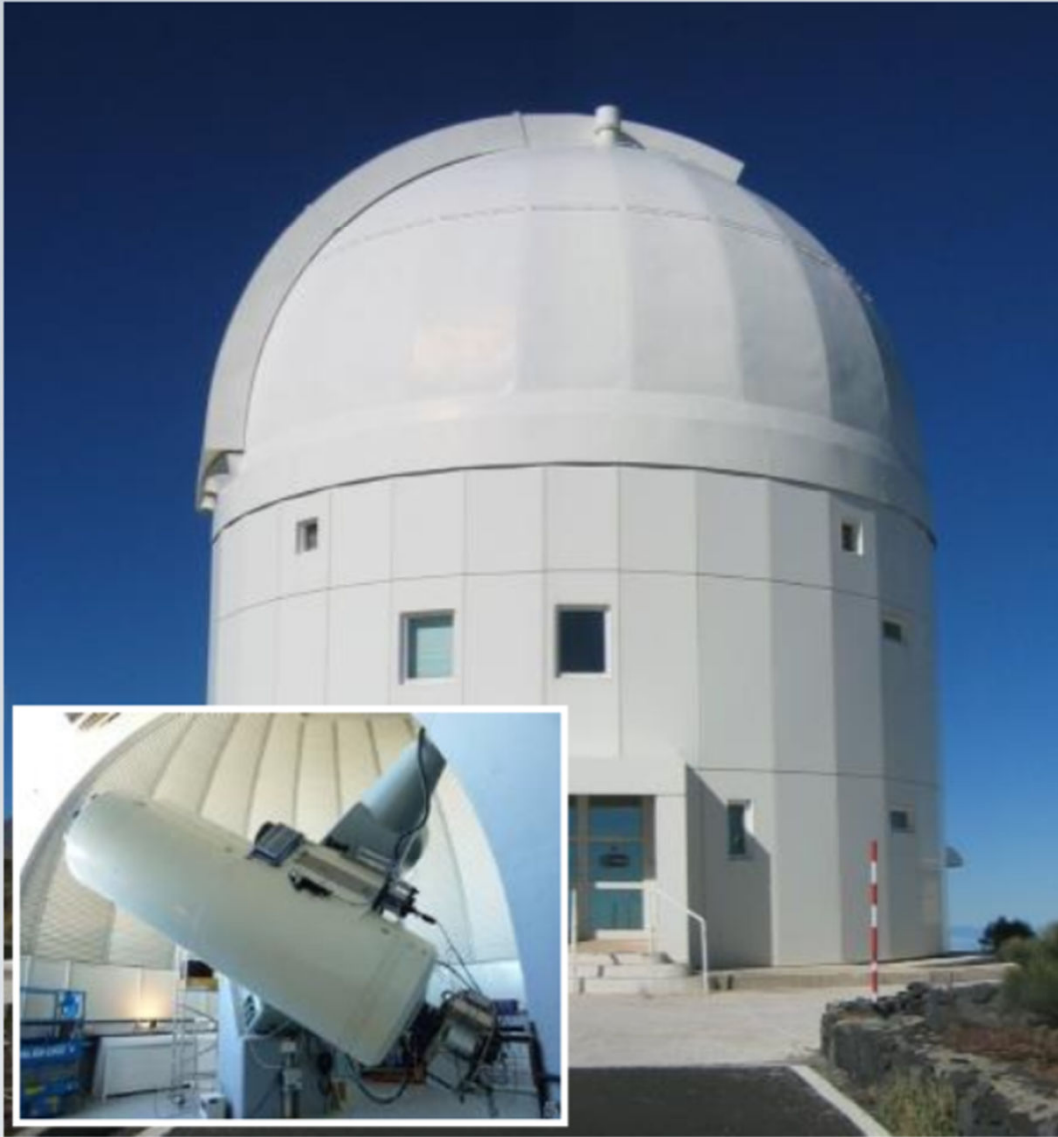
Optical LEO – Direct-to-Earth: Measured Received **Power** and Bit Error Rate over Elevation



Optical Ground Stations



ESA-OGS (Teneriffa), 1 m



Bilder: ESA

Some more Ground Stations



http://global.jaxa.jp/press/2006/04/img/20060407_kirari_pic01.jpg

- OGS des NICT
- Standort: Tokyo
- 1,5 m



http://global.jaxa.jp/press/2006/04/img/20060407_kirari_pic01.jpg

- OGS von NASA-JPL
- Standort: Table Mountain, Kalifornien
- 1 m

DLR Ground Stations



Station	Location	Operational Readiness	Aperture	Uplink Beacon (baseline)
FOGATA	Tabernas, Almeria	Q2-2024	600-mm	1590-nm
LaBoT	Trauen, Fassberg	Q2-2024	700-mm	1590-nm
OGSOP	Oberpfaffenhofen, Munich	Running	800-mm	various
OGSTB	Oberpfaffenhofen, Munich	Q1-2024	304-mm	1590-nm, 1560-nm, 967-nm
TOGS	Mobile	Running	600-mm	various

Optical Ground Station Oberpfaffenhofen „Next generation“ – OGSOP-NG



Improved performance and sensitivity

- 80 cm aperture
- Measurements with better spatial resolution
- Supports links in GEO-, deep space- and quantum key distribution-applications

Multiple foci, including Coudé

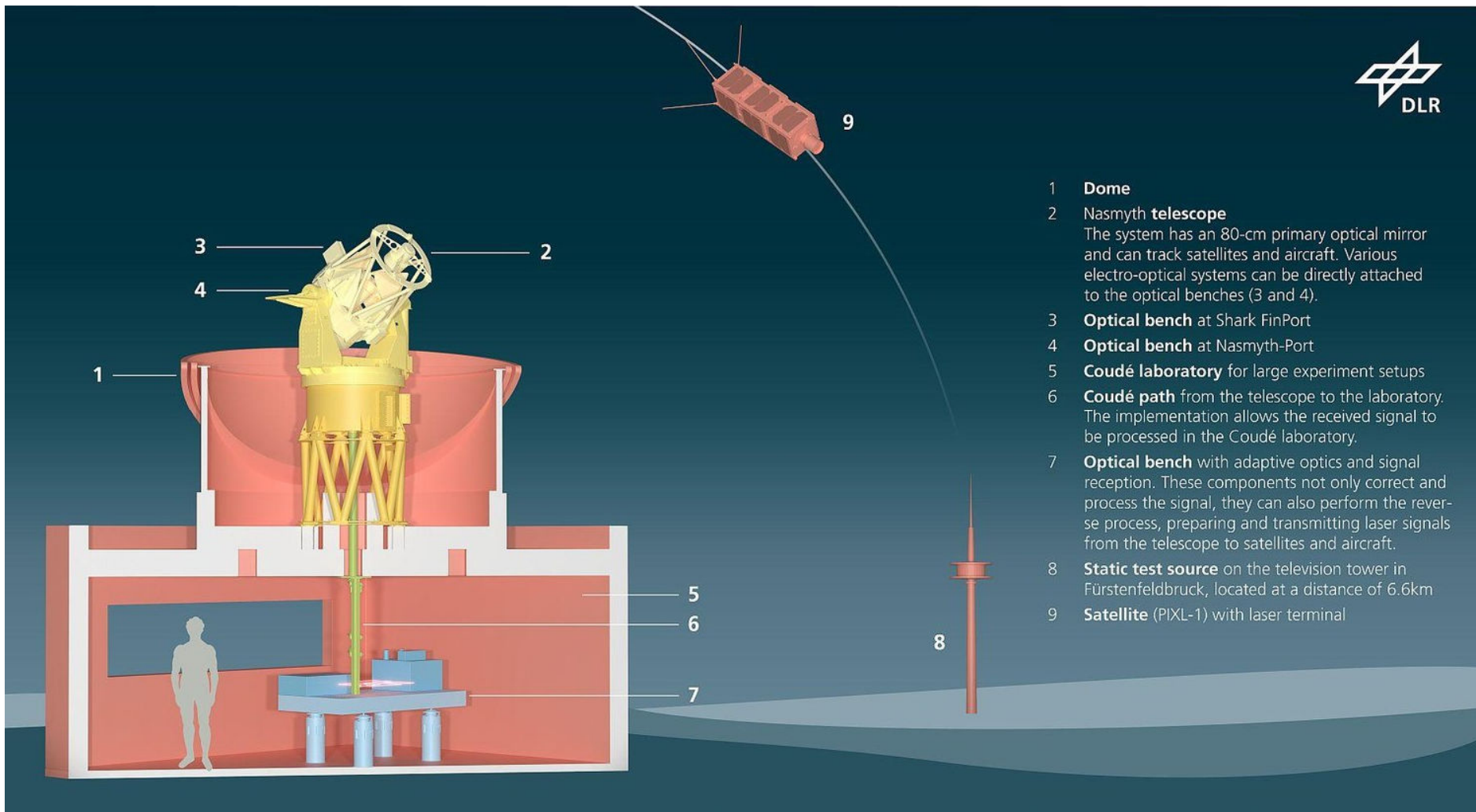
- High flexibility to change between setups, enabling multi-mission support
- Adaptive Optics on Coudé-Bench

Characterization of the atmosphere

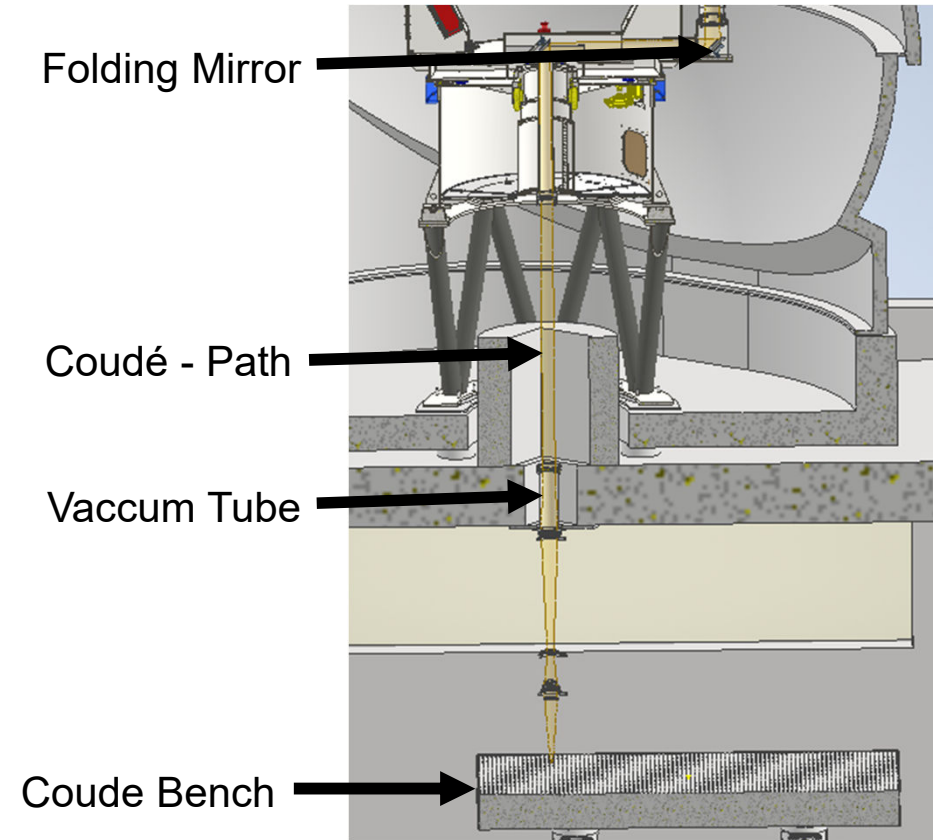
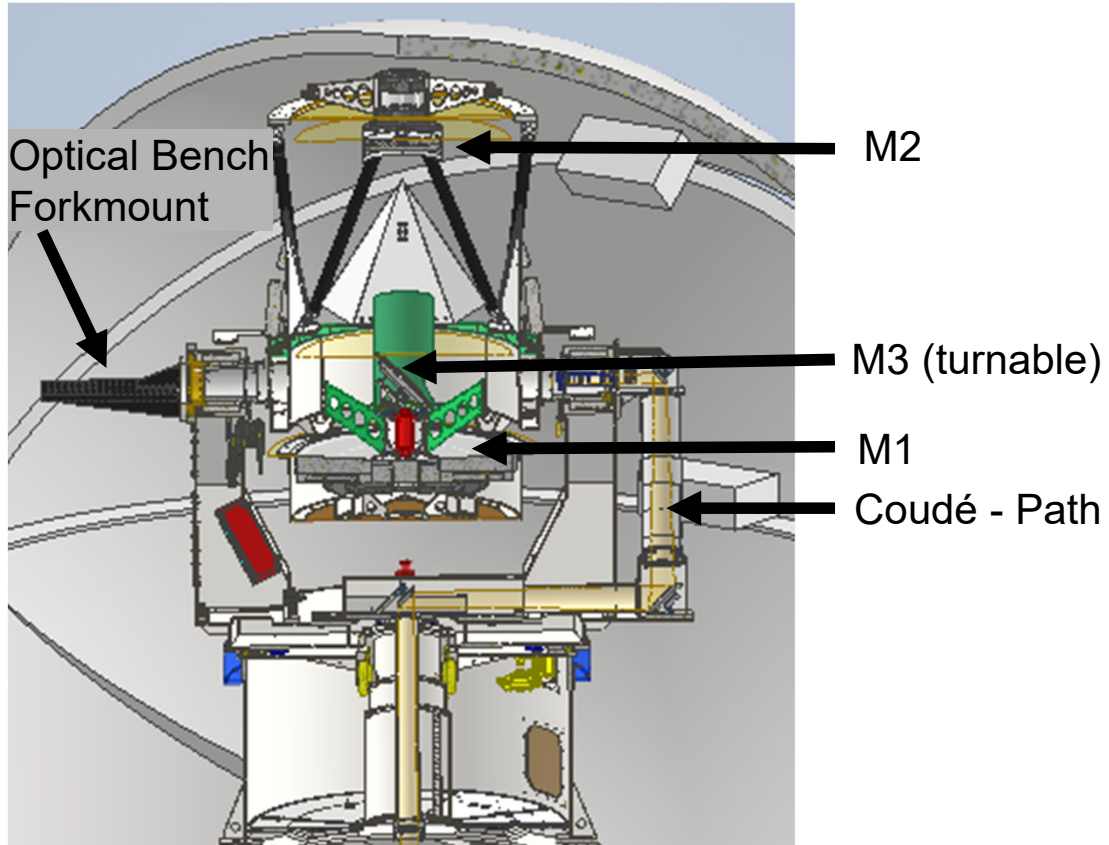
- Measurement instruments for recording of key atmospheric parameters



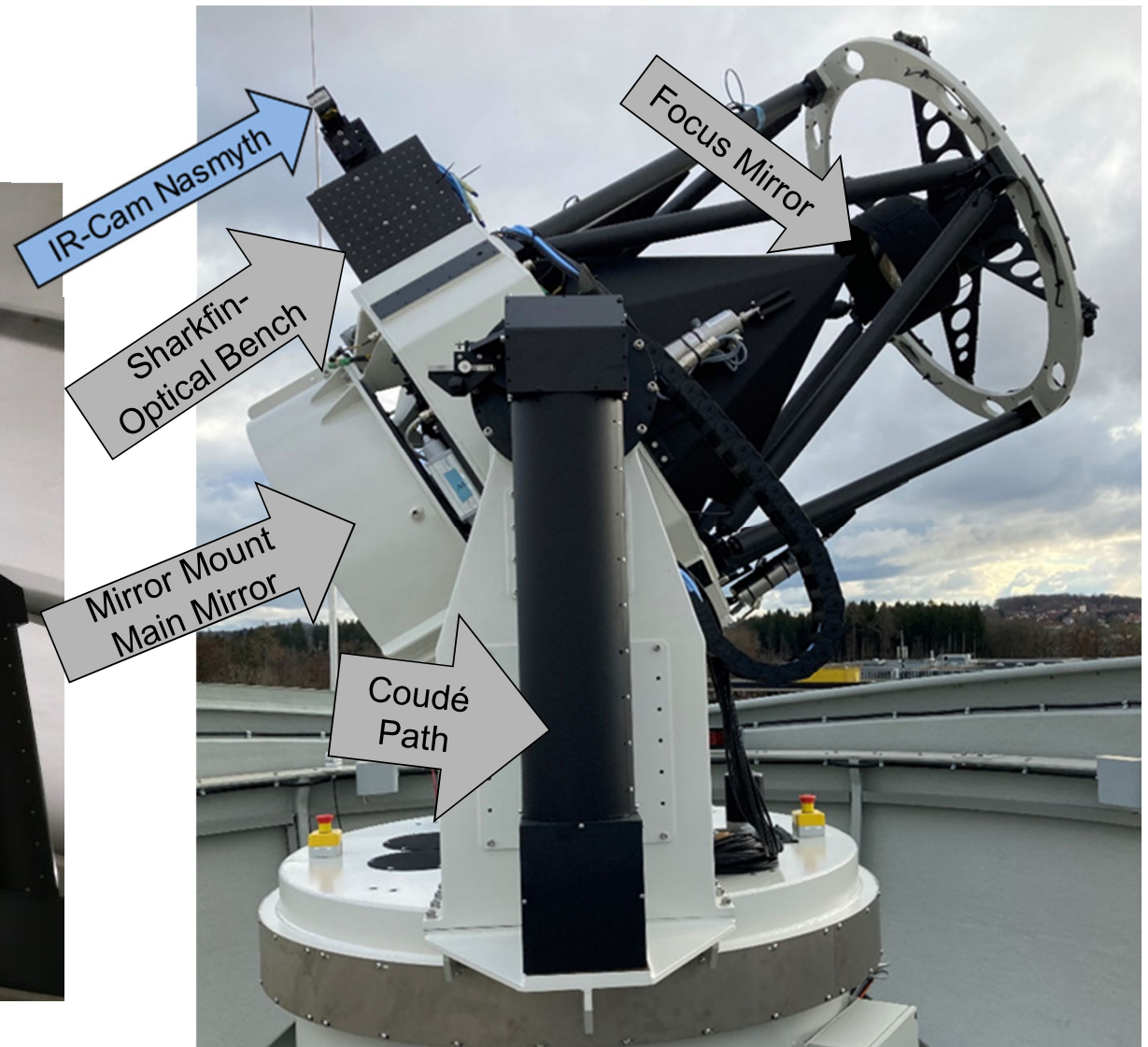
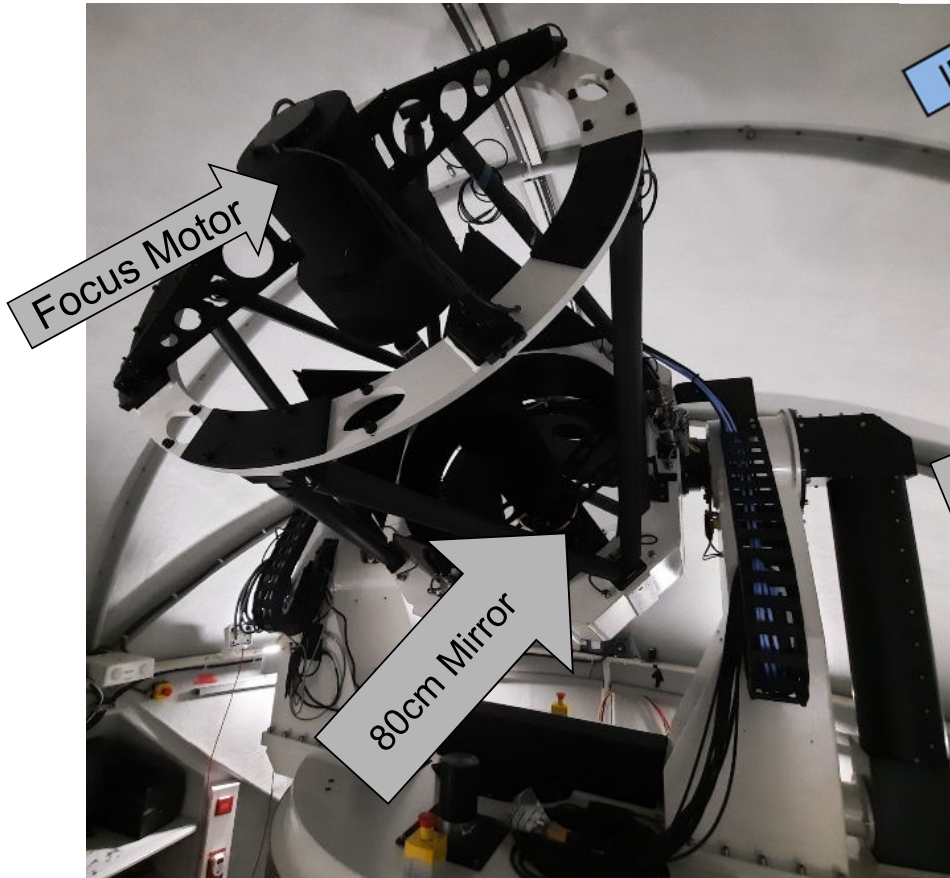
Development OGSOP-NG



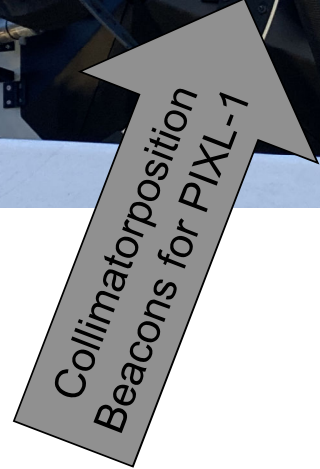
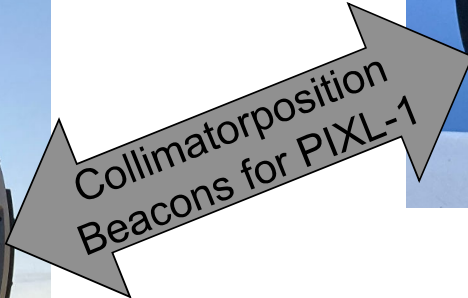
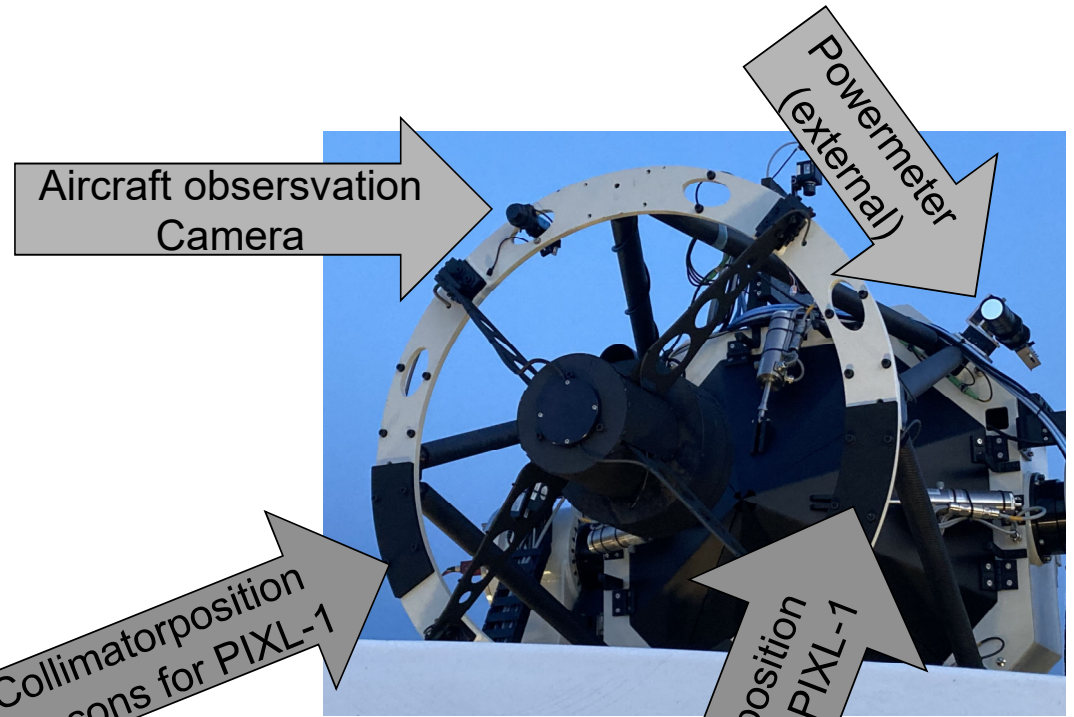
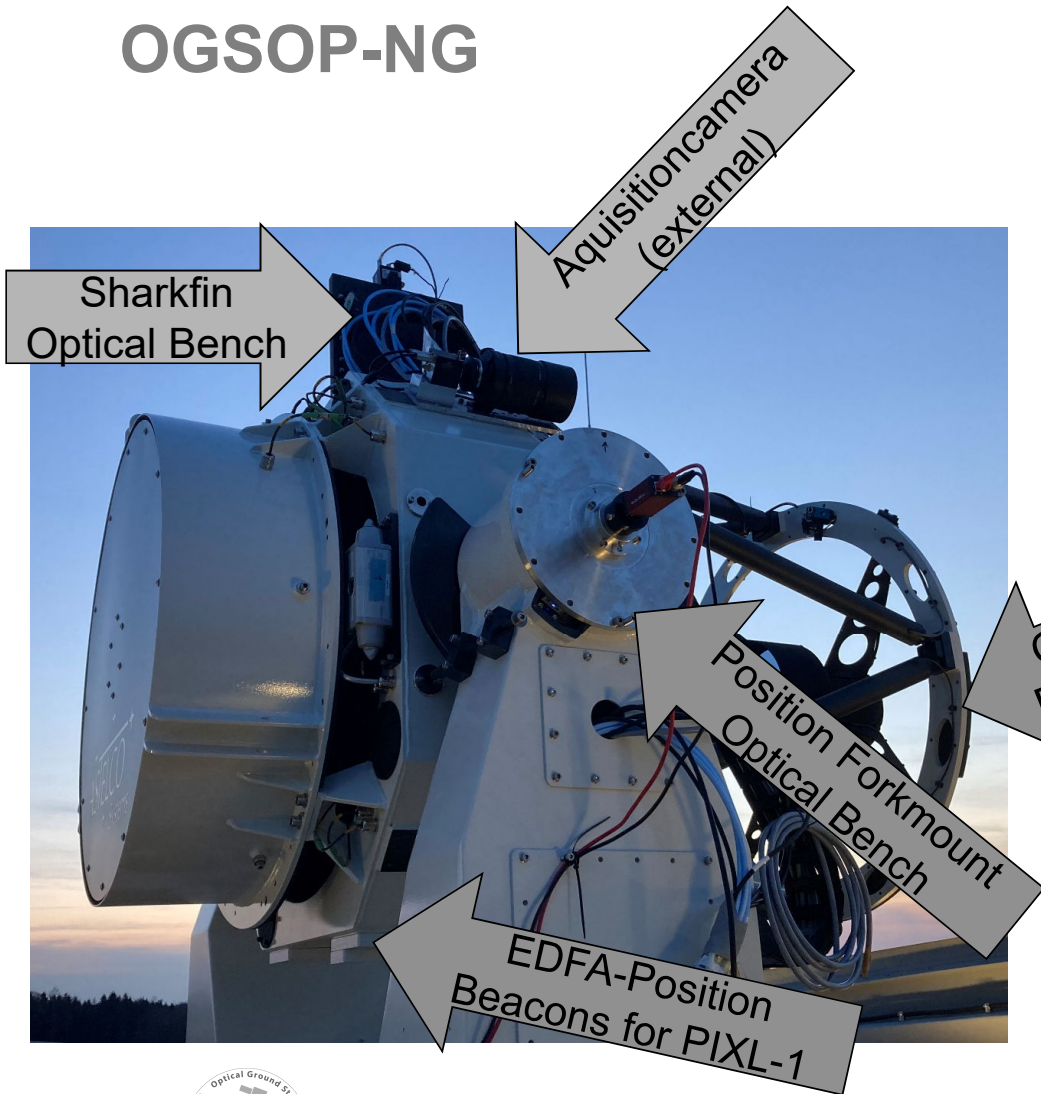
Development OGSOP-NG



OGSOP-NG



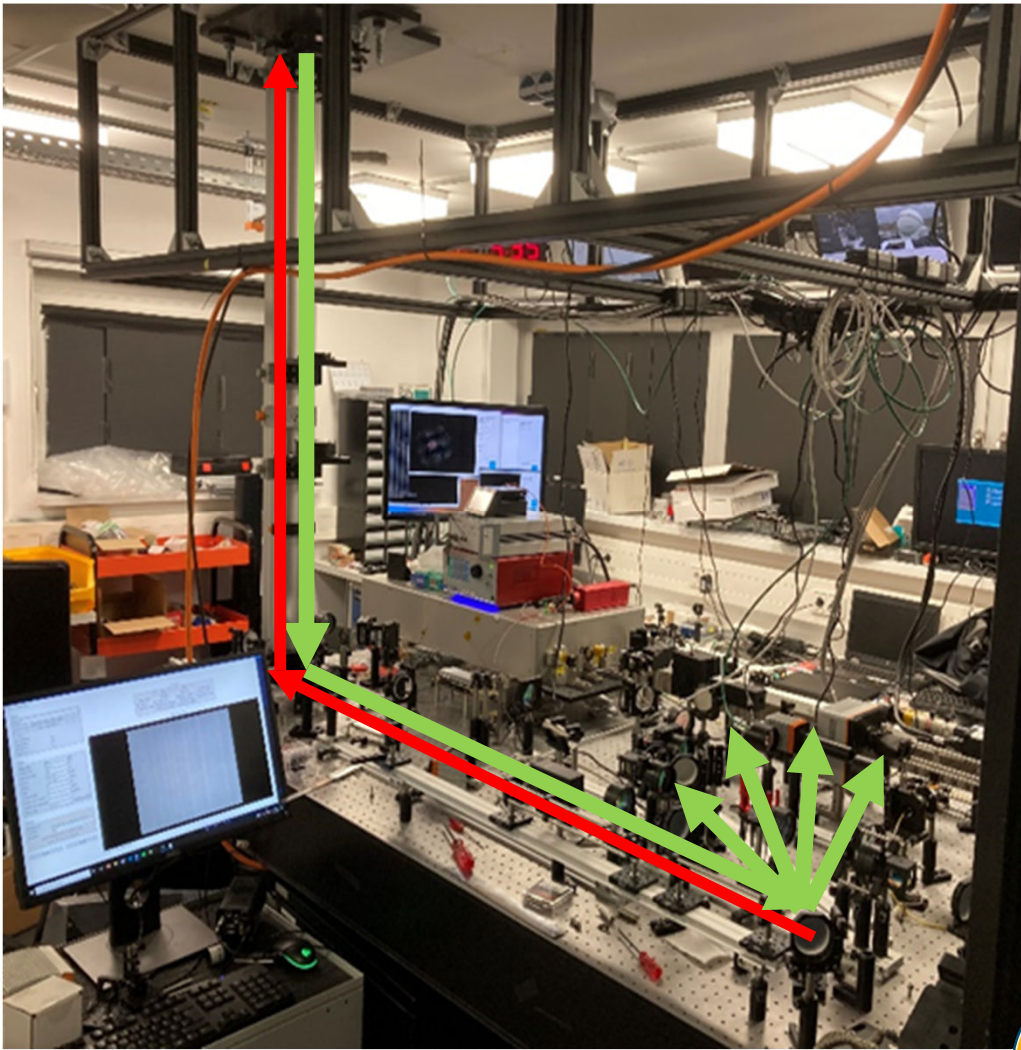
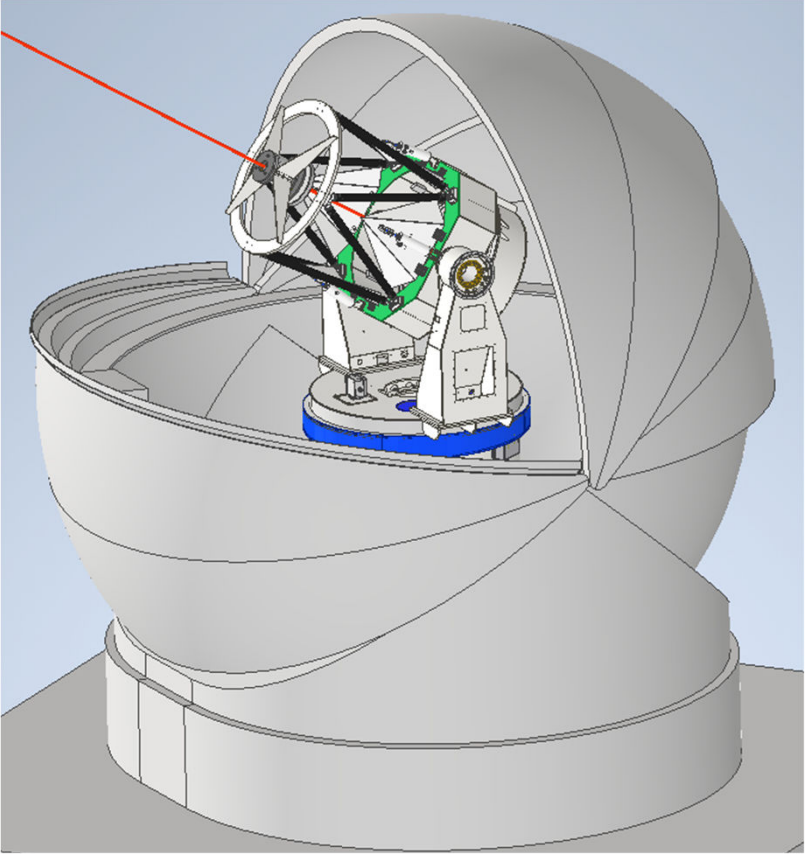
OGSOP-NG



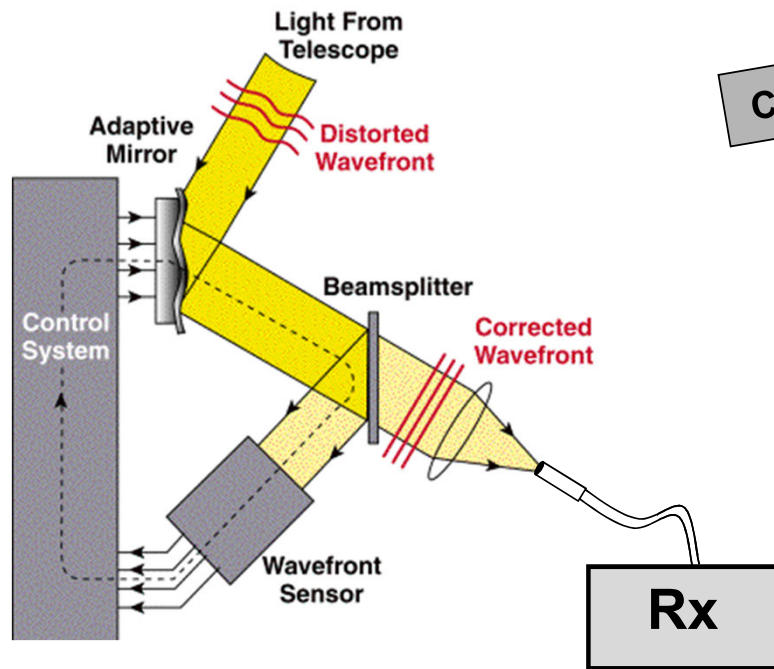
- **Sharkfin** optical Bench
300mm X 300mm
- **Forkmount** Optical Bench
500mm X 500mm
- Several smaller **Side installation spaces**
Parallel to telescope axis



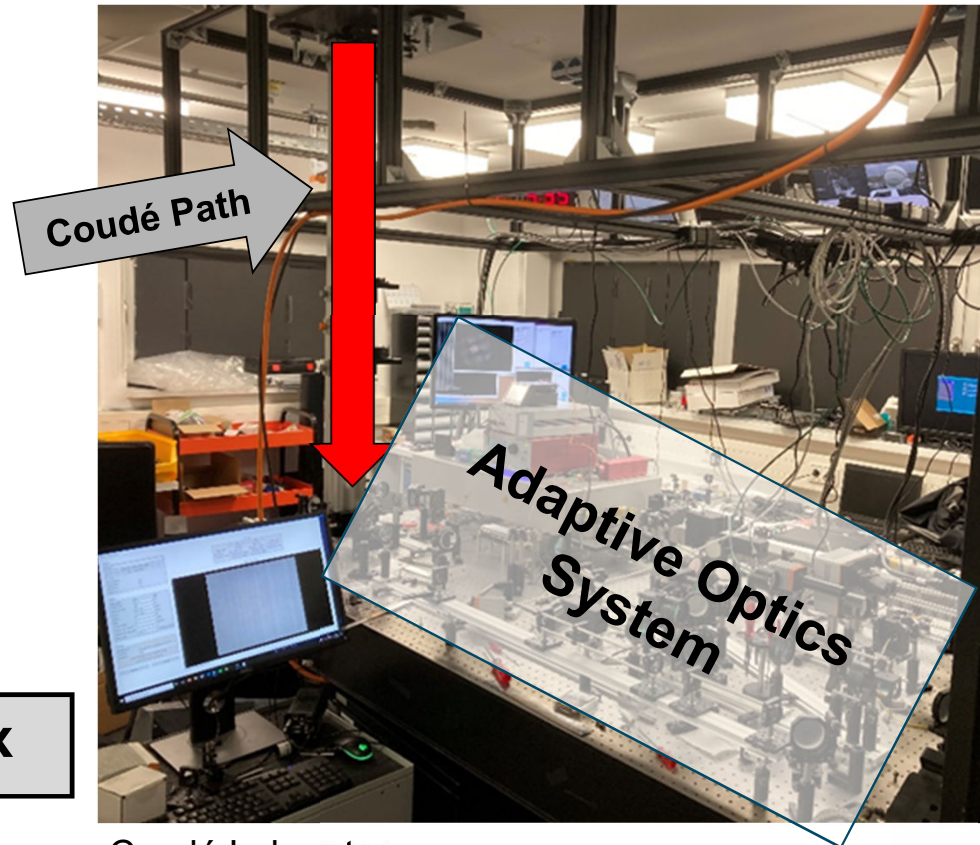
Coudé-Bench



OGSOP – Coudé room with adaptive optics



AO System Concept



Coudé Laboratory

Meas. with
Alphasat-LCT

Without AO

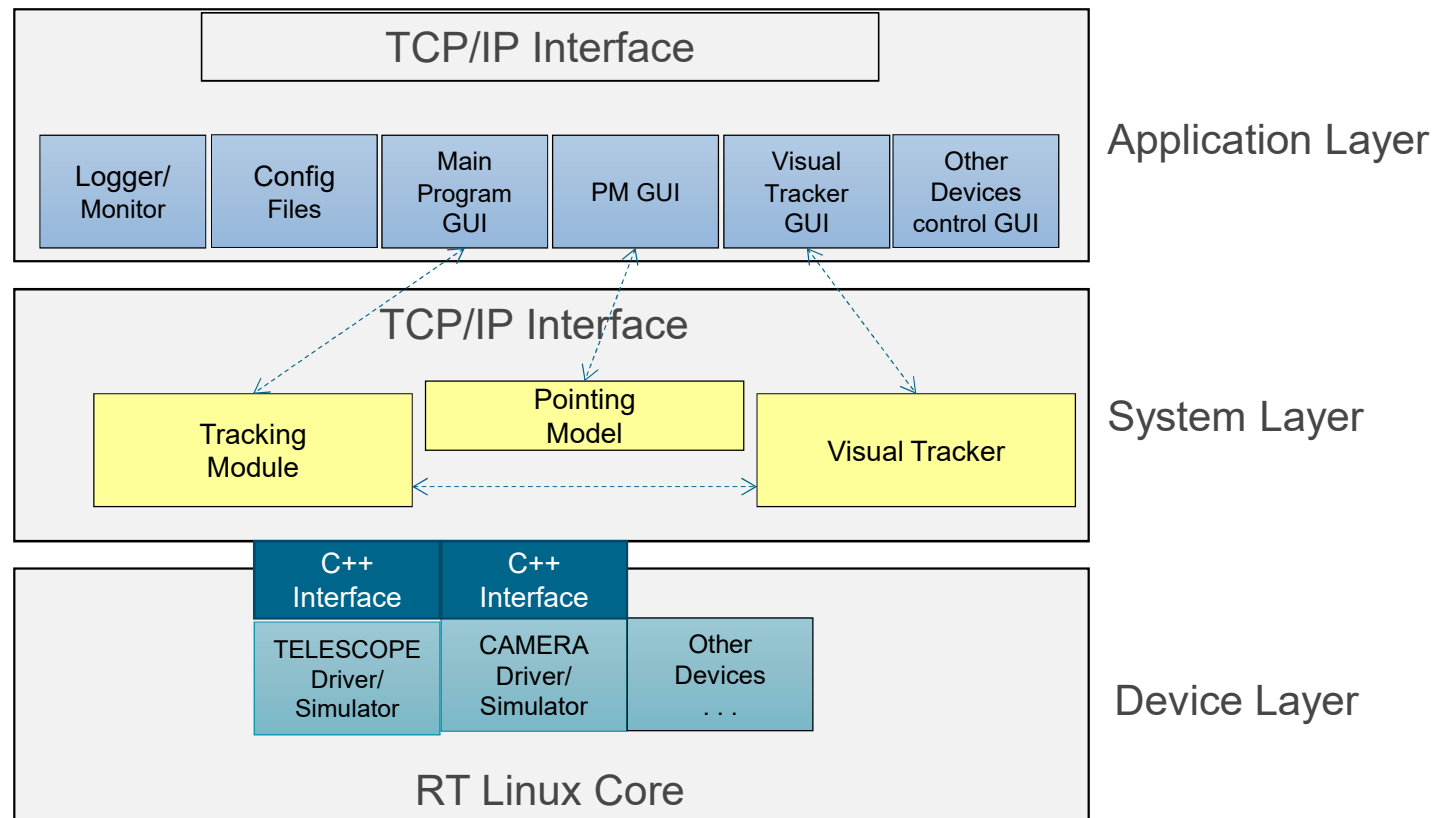


With AO

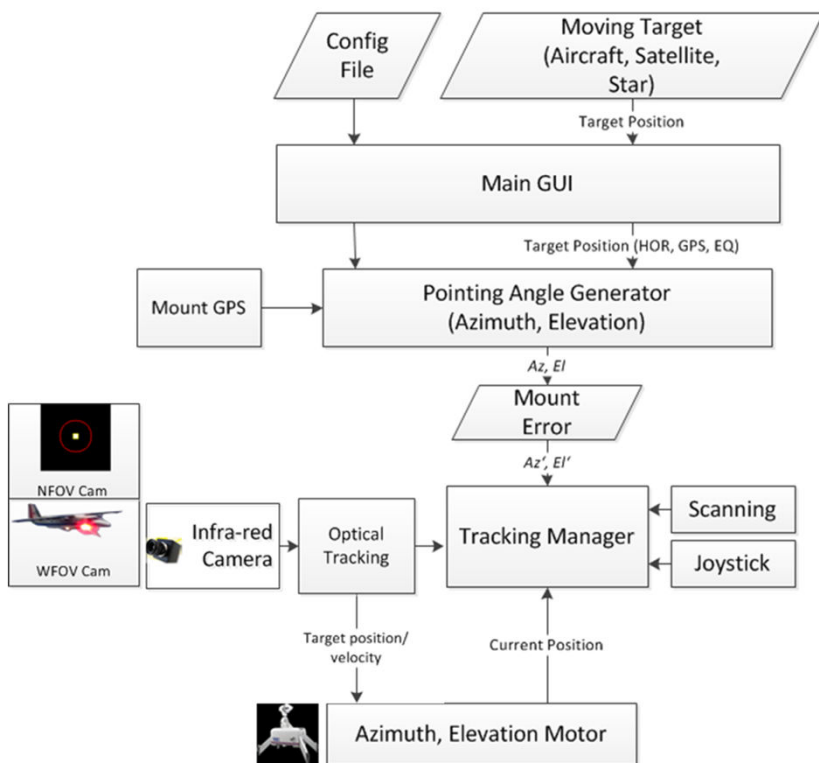


Gefördert durch
Bayerisches Staatsministerium für
Wirtschaft und Medien, Energie
und Technologie

OGS Software Overview



Software Main Features



GPS Tracking

- GPS target (Dynamic or static)
- GPS extrapolation

Satellite Tracking

- Two line elements (TLE)
- Consolidated Prediction Format (CPF)
- Orbit Ephemeris Message (OEM)

Star Tracking

- Star catalogue
- Right ascension (RA), Declination (DEC)

Open Loop Tracking

Optical Tracking

Software User Interface



Main GUI

File Control

Scanning

Velocity: 30,00000 */s U: 0,0000 *

Increment: 0,028000 */round V: 0,0000 *

Radius U: 20,000000 * Radius V: 20,000000 *

Manual

Fixed

U: 0,0000 * V: 0,0000 *

Joystick

U: 0,0000 * V: 0,0000 * 1,000000 */s

Time offset: 0,00 ms

Camera

dU: 0,000000 * U: 0,000000 * Camera Only

dV: 0,000000 * V: 0,000000 * Reset Total

Mount

Instrumental

U: 126.858102 * Az: 126.858102 *
V: 5.727761 * El: 5.727761 *

State

U: Tracking V: Tracking

Velocity

U: -0.019869 * V: -0.082763 *

Target Position

Satellite: [Dropdown]

Select Satellite: Starlink 2210

TLE

Line 1: 1 47777C 21017BH 22025.14600593 .00007246 00000-0 48578-3 0 256
Line 2: 2 47777 53.0530 125.7894 0001575 62.3503 145.3764 15.06367369 10

Time

Universal Time: 07:52:58 Local Time: 08:52:58 Julian Date: 2459604.82846

Horizontal Coordinates

Azimuth: 126.858088 * Elevation: 5.727723 *

Target Position

Az: 126.858288 * El: 5.728554 *
U: 126.858288 * V: 5.728554 *

Own Position

Fixed: [Dropdown]

GPS Target Position

Latitude: 48,084680 * Longitude: 11,278150 * Height: 594,000000 m

IMU

Yaw: 0,000000 * Pitch: 0,000000 * Roll: 0,000000 *

Visual Tracker

Visual Tracker

Original Image | Processed Image

Camera - 100 fps

IntegrationTime: 0 ms

High Gain Rotation: 0.00 *

Target Position

X: 160 px Y: 128 px

Inner Circle α : 1.00 mrad Outer Circle α : 2.00 mrad

Spot

dX: 135 px dY: 140 px

α : -0.0215 * β : 0.0109 *

Spot Parameters

Minimum size: 2 Maximum size: 50 Threshold: 128

Visualization

Image Display Rate: 100 % Auto Background Color Store Images

Rotation Calibration

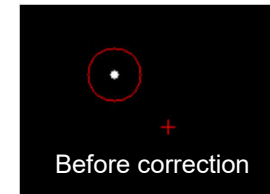
Point 1: [Dropdown] Point 2: [Dropdown]

Rotation: 0.00 * Angle: 0.00 *

Statistics

Min Cycle Time: 2.31 ms Max Cycle Time: 4.51 ms

Calibration



Pointing model is calculated with the help

- Simple Pointing Model
- Classic Pointing Model
- Extended Pointing Model

Requirement for good calibration:

- Enough star measurements
- Uniform distribution of targets in the sky
- Accurate timing of the system

OGS Pointing Model GUI

Calculation Measurement

Reload

	ID	Name	RA 2000	Dec 2000	Az	Elv	Magnitude
1	HR 15	Alp And	0:08:23.30	29:05:26.0	57:35:25.8	10:36:44.7	2.1
2	HR 21	Bet Cas	0:09:10.70	59:08:59.0	34:59:58.3	32:13:49.6	2.3
3	HR 153	Zet Cas	0:36:58.30	53:53:49.0	36:11:27.9	25:49:33.4	3.7
4	HR 165	Del And	0:39:19.70	30:51:39.0	51:10:57.0	7:48:39.0	3.3
5	HR 168	Alp Cas	0:40:30.50	56:32:14.0	33:47:02.1	27:26:04.2	2.2
6	HR 219	Eta Cas	0:49:06.00	57:48:57.0	31:50:37.7	27:36:08.1	3.4
7	HR 264	Gam Cas	0:56:42.50	60:43:00.0	28:49:53.4	29:07:37.6	2.5
8	HR 269	Mu And	0:56:45.20	38:29:58.0	43:37:55.4	11:45:01.9	3.9
9	HR 337	Bet And	1:09:43.90	35:37:14.0	43:10:19.3	7:56:11.7	2.1
10	HR 403	Del Cas	1:25:49.00	60:14:07.0	26:01:16.0	26:30:12.5	2.7
11	HR 424	Alp UMi	2:31:48.70	89:15:51.0	0:27:52.5	47:32:30.2	2.0
12	HR 464		1:37:59.60	48:37:42.0	31:25:59.9	15:55:17.9	3.6

Telescope Control

Stop Track Acquire Measurement

Measurement

Status

Time

UT :022-01-25 07:53:53

LocT :022-01-25 08:53:53

JD 0.00000

SidT 0:00:00

WGS 84 Position

Lon 0:00:00.0 Alt 0.0

Lat 0:00:00.0

STOPPED

Equatorial Coordinates

RA 0:00:00.00 Dec 0.0

HA 0.0 ParA 0.0

Horizontal Coordinates

Az 0.0 Elv 0.0

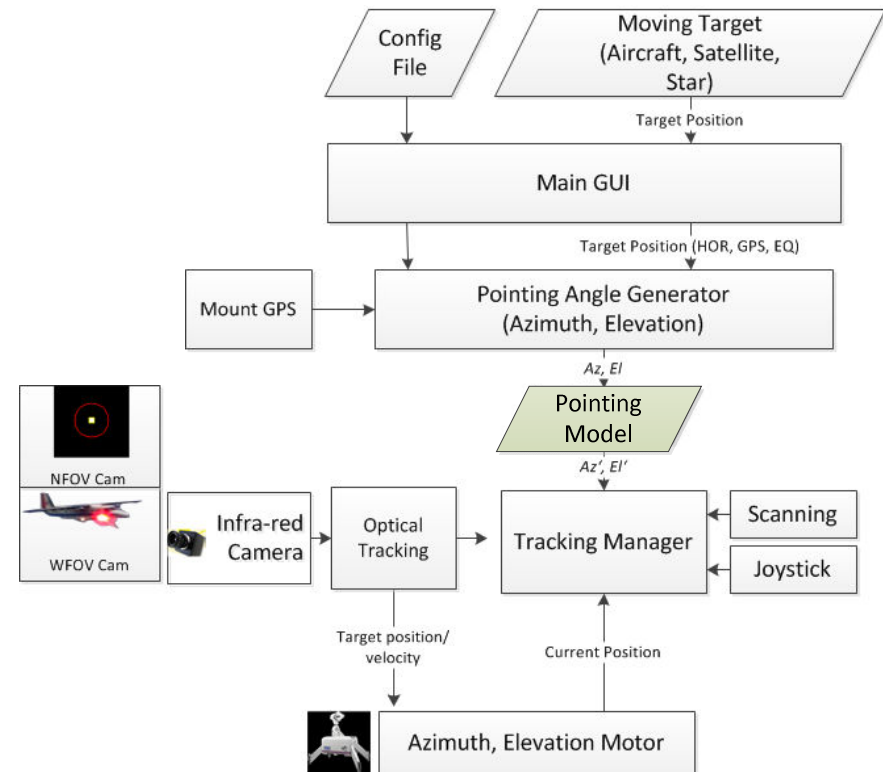
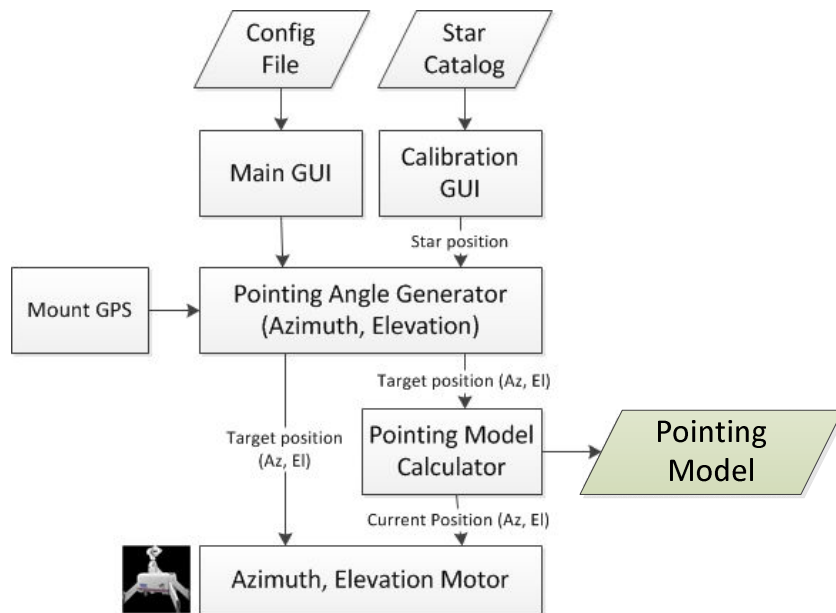
Instrumental Coordinates

U 0.0 UOff 0.0

V 0.0 VOff 0.0

Tracking accuracy up to $10\mu\text{rad}$ has been achieved as a result of good calibration

Pointing Model



Satellite Based QKD

Why Satellite?

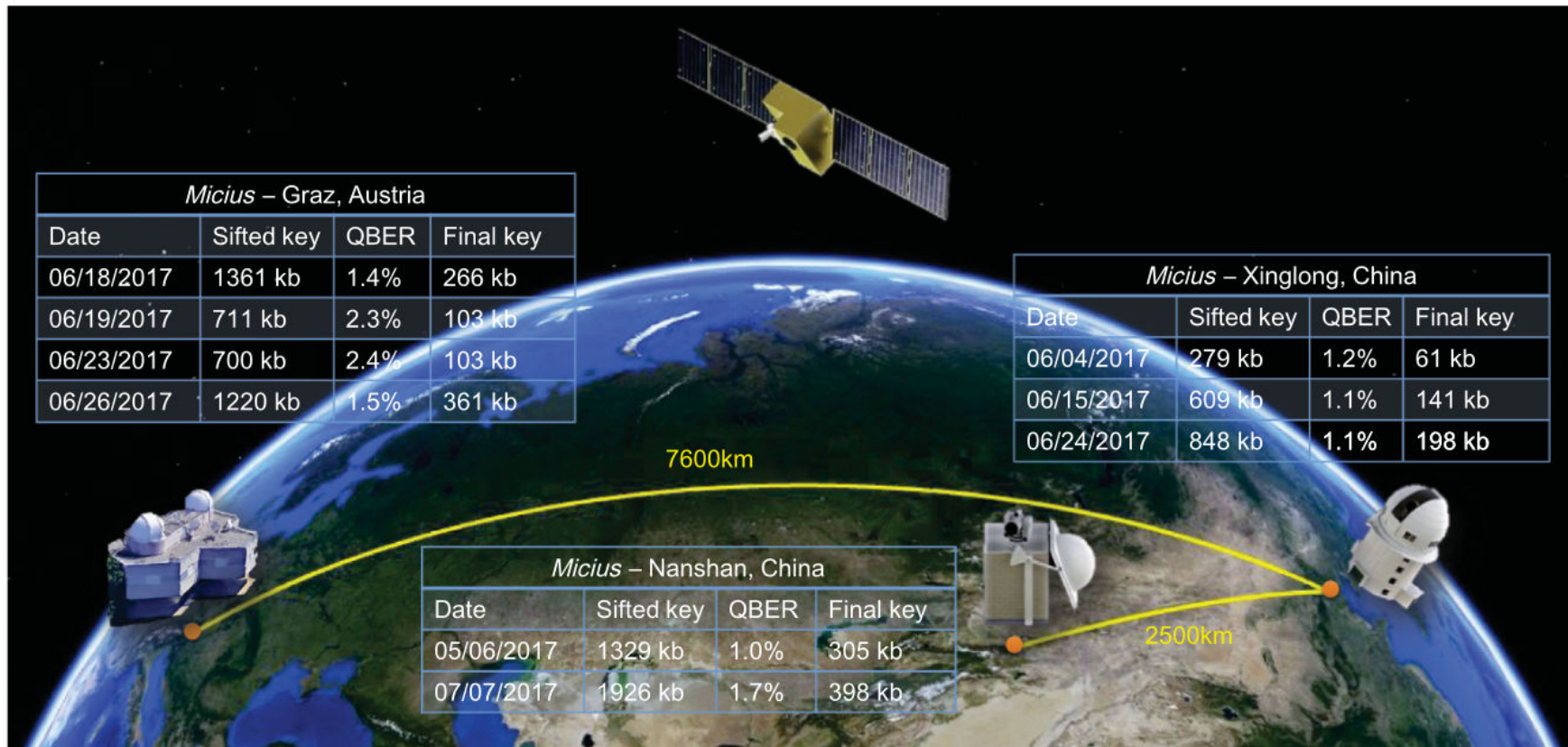
→ bridge large distances

- **FSO and QKD** enable worldwide **fast** and **secure** data communications
- Why **QKD over FSO**? → use of fiber is range limited → QKD using satellite node with QKD relay protocol
- Combination of FSO and QKD technology in one device → high synergies → lower costs
- QKD schemes: Discrete variable (BB84) or continuous variable (and various others..)



BB84 Story

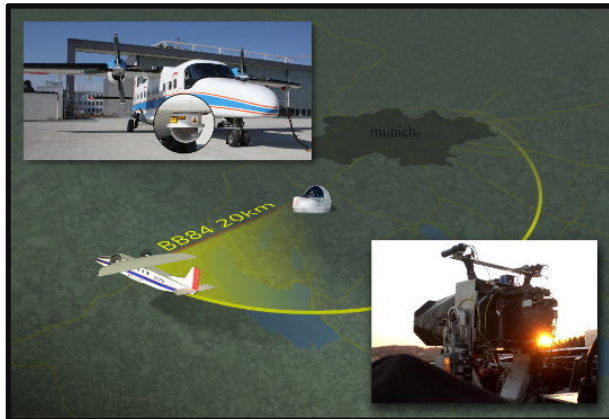
Satellite Experiment with Micius (since 2016)



Liao et al. Satellite-Relayed Intercontinental Quantum Network Phys. Rev. Lett., American Physical Society, 2018, 120, 030501

QKD Projects

Aircraft QKD



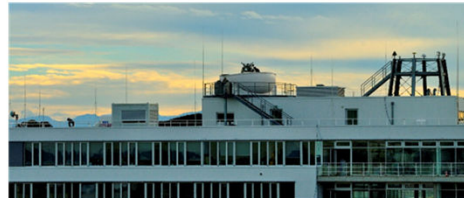
QuKomIn

Funkturm FFB

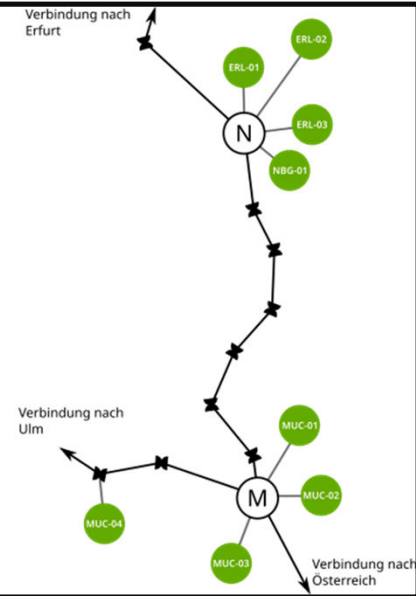


Quantenlabor

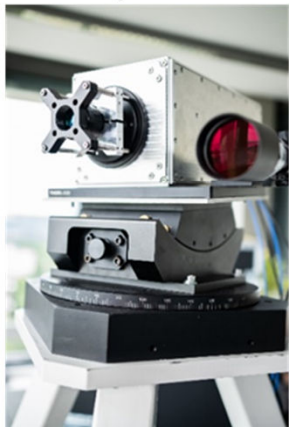
IKN roof with OGS



Optiklabor



Tx/Rx pair for satellite QKD Demo



Quantum secure video conference BMBF/BSI



Further Reading and Lectures:



- S.G. Lambert, W.L. Casey, "Laser Communications in Space", Artech House, 1995
- B.J. Klein, J.J. Degnan, „Optical Antenna Gain. 1: Transmitting Antennas“, Applied Optics Vol. 13(9), 1974
- W.K. Pratt, “Laser Communication Systems“, Wiley publishing, 1969
- L.C. Andrews, R.L. Phillips, ”Laser beam propagation through random media“, SPIE-Press 2005
- D. Giggenbach, F. Moll, C. Fuchs, C. Schmidt, A. Shrestha, “*Optical on–off keying data links for low Earth orbit downlink applications*“, Chapter in ‘Satellite Communications in the 5G Era’, Editors S. K. Sharma, S. Chatzinotas, P-D Arapoglou, IET TELECOMMUNICATIONS SERIES, 2018
- D. Giggenbach, F. Moll, „*Scintillation Loss in Optical Low Earth Orbit Data Downlinks with Avalanche Photodiode Receivers*“, IEEE-Xplore, Int. Conf. on Space Optical Systems 2017 (ICSOS), 2017
- D. Giggenbach, A. Shrestha, “Atmospheric Absorption and Scattering Impact on Optical Satellite-Ground Links” Int. Jnl. of Satellite Communications And Networking, 2021

- *S. Scalise, T. de Cola, „Satellite Communications“, yearly lecture at TUM, Winter Semester*
- *M. T. Knopp, „Optical Communications“, yearly lecture at UniBW-Munich*
- *D. Giggenbach, N. Hanik, C. Fuchs, R. Mata-Calvo, “Optische Kommunikation für Satelliten und Flugzeuge“, yearly course at Carl-Cranz-Gesellschaft CCG, Wessling*

Impressum



Topic: **Optical Satellite Links at DLR**
Overview about the Optical Satellite Links department at DLR's
Institute of Communications and Navigation

Date: 10.09.2024

Authors: Amita Shrestha (amita.shrestha@dlr.de) et. Al.

Institute: Institute of Communications and Navigation

Pic.-credits: DLR

